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Design and Implementation of a X-band Transmitter and Frequency Distribution Unit for a Synthetic Aperture Radar

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A dissertation submitted to the Department of Electrical Engineering,
University of Cape Town, in fulfilment of the requirements
for the degree of Master of Science in Engineering.

Cape Town, May 2004

Declaration

I declare that this dissertation is my own, unaided work. It is being submitted for the degree of Master of Science in Engineering at the University of Cape Town. It has not been submitted before for any degree or examination in any other university.

Signature of Author

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Abstract

Synthetic aperture radar (SAR) can provide high-resolution images of extensive areas of the earth's surface from a platform operating at long ranges, despite adverse weather conditions or darkness. A local consortium was established to demonstrate a consolidated South African SAR ability to demonstrate to the local and international communities, by generating high quality images with a South African X-band demonstrator. This dissertation forms part of the project. It aims to describe the design and implementation of the transmitter and associated frequency distribution unit (FDU) for the SASAR II, X-band SAR.

Although the transmitter and FDU are two separate units, they are ultimately linked. The transmitter has the task of taking a low-power, baseband, chirp waveform and, through a series of mixers, filters and amplifiers, converting it to a high-power, microwave signal. The FDU is essentially the heart of the transceiver and provides drive to all the mixer local oscillator (LO) inputs. It also clocks the DAC and ADC which allow the essentially analogue transceiver to communicate with the digital circuitry.

It is found that the chirp signal produced is of satisfactory fidelity. LO feedthrough, however, is superimposed at the chirps' centre frequency. As a result of previous stages, spurious signals exist at 16 MHz offset from the chirps' centre frequency and at 9142 MHz. The system transfer function reveals that 2 dB roll-off is present at the outer frequencies of the chirp signal. Group delay in the transmitter filters and amplifiers is held responsible for this.

Acknowledgements

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I would also like to thank Richard Lord and Thomas Bennett, who were always there for me, whenever help was needed. The rest of the Radar Remote Sensing Group (RRSG) who put up with me over the past two years also deserve a special mention. Without them, it would have been a far less enjoyable experience.

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List of Symbols

a	—	Waveguide horizontal length [cm]
A	—	Amplitude [m]
b	—	Waveguide vertical length [cm]
B	—	RF bandwidth [Hz]
c	—	Speed of light [m/s]
C	—	Carrier signal power [W]
E_{max}	—	Maximum field strength [V/cm]
f_m	—	Frequency offset from carrier [Hz]
f_0	—	Carrier frequency [Hz]
h	—	Height [m]
K	—	Chirp rate [Hz/s]
l	—	Length [m]
N	—	Noise power in 1-Hz bandwidth at f_m offset from carrier [W]
P	—	Power [W]
S	—	FGCW centre conductor width [m]
t	—	Time [s]
T	—	Pulse length [s]
V_{p-p}	—	Voltage peak-to-peak
w	—	Width [m]
W	—	FGCW inner spacing width [m]
ϵ	—	Permittivity [F/m]
ϵ_0	—	Absolute constant of permittivity [F/m]
ϵ_r	—	Relative permittivity or dielectric constant
λ	—	Wavelength [m]
λ_g	—	Guide wavelength [m]
ω	—	Angular frequency [rad/s]

Nomenclature

ADC—Analogue to Digital Converter.

AWG—Arbitrary Waveform Generator.

Azimuth—Angle in a horizontal plane, relative to a fixed reference, usually north or the longitudinal reference axis of the aircraft or satellite.

BIT—Built in Test.

Chirp—A pulse modulation method used for pulse compression. The frequency of each pulse is increased or decreased at a constant rate throughout the length of the pulse.

Coherence—A continuity or consistency in the phases of successive radar pulses.

CW—Continuous Wave.

DAC—Digital to Analogue Converter.

Doppler frequency—A shift in the radio frequency of the return from a target or other object as a result of the object's radial motion relative to the radar.

DPG—Digital Pulse Generator.

DSU—Data Storage Unit.

FDU—Frequency Distribution Unit.

FGCW—Finite Ground Coplanar Waveguide.

GPS—Global Positioning System.

I—In-phase.

IF—Intermediate Frequency.

LO—Local Oscillator.

Nadir—The point directly below the radar platform.

NAV—Navigation Unit.

PA—Power Amplifier.

PCB—Printed Circuit Board.

PLL—Phase-Locked Loop.

PRF—Pulse Repetition Frequency.

Q—Quadrature.

Radar—Radio Detection and Ranging.

Range—The radial distance from a radar to a target.

RCU—Radar Controller Unit.

RDU—Radar Digital Unit.

RF—Radio Frequency.

RFU—Radio Frequency Unit.

RRSG—Radar Remote Sensing Group (UCT).

SASAR II—South African Synthetic Aperture Radar.

Synthetic Aperture Radar (SAR)—A signal processing technique.

Swath—The area on earth covered by the antenna signal.

TWT—Travelling Wave Tube.

UCT—University of Cape Town (South Africa).

VCO—Voltage Controlled Oscillator.

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Chapter 1

Introduction

This dissertation describes the design and implementation of the transmitter and associated frequency distribution unit (FDU) for the SASAR II, X-band, synthetic aperture radar (SAR). Various local companies provided the impetus for this project to be launched with the aim of consolidating and developing local SAR knowledge.

SAR can provide high-resolution images of extensive areas of the earth's surface from a platform operating at long ranges, despite adverse weather conditions or darkness. This resistance to weather stems from the use of wavelengths of the order of centimetres. X-band (3cm), C-band (6cm) and L-band (24cm) are the favoured frequency choices [1].

Although the transmitter and FDU are two separate units, they are ultimately linked as can be seen in Figure 1.1. The transmitter has the task of taking a low-power, baseband waveform and converting it to a high-power, microwave signal. The FDU is essentially the heart of the transceiver and provides drive to all the mixer local oscillator (LO) inputs. It also clocks the DAC and ADC which allow the essentially analogue transceiver to communicate with the digital circuitry.

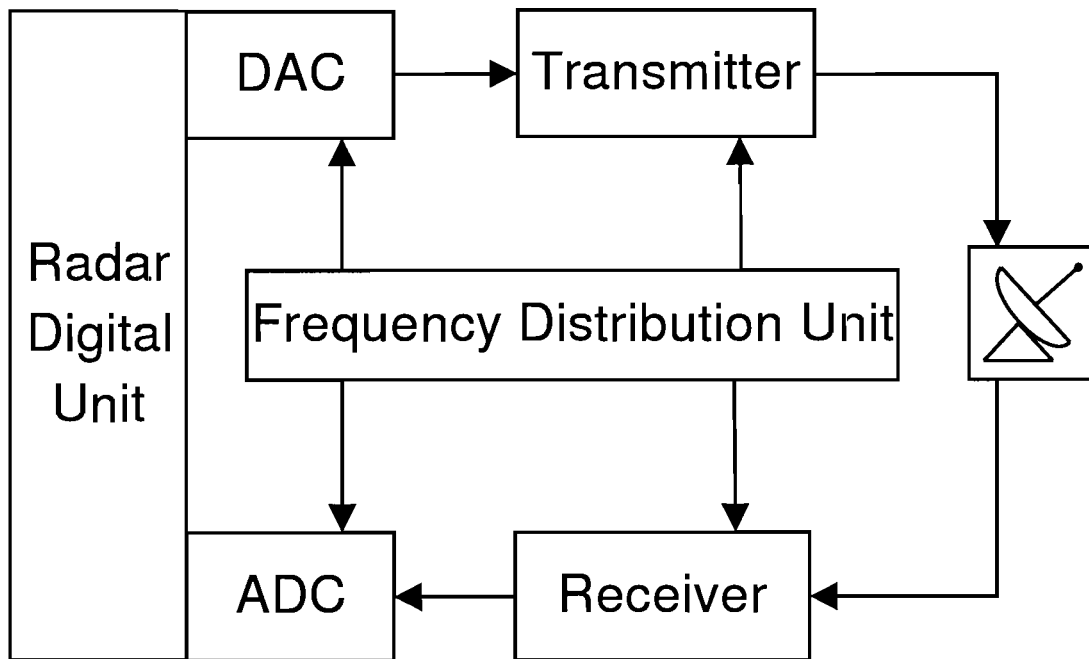


Figure 1.1: Block diagram showing how the transceiver interfaces with the FDU and Radar Digital Unit.

The information required for this project has been gathered from relevant books, the Internet and from discussions with various people in the radar and associated communities. Constraints on the project have been excessive red-tape and a lack (or delay) of funding.

1.1 Terms of Reference

The following terms of reference were agreed upon by the SASAR II system engineer:

- The DAC shall output two channels at maximum power of +10 dBm each. Each baseband channel shall have bandwidth of 50 MHz.
- The system shall have two IF stages at 158 MHz and 1300 MHz and a RF stage at 9300 MHz.
- System filtering will comprise of 5th order Butterworth filters.
- The transmitted peak power level shall be 3.5 kW.
- The system shall have a pulse repetition frequency (PRF) of 3 kHz and pulse length of 5 μ s.
- The DAC and ADC shall be clocked at 150 MHz and 210 MHz respectively. The clock jitter shall not exceed 6 ps.
- The system shall have a set of states and modes for testing instructions.

1.2 Plan of Development

This document is arranged as follows:

Chapter 2: Background Information

The following chapter gives the background to this dissertation. It begins by exposing the reasons for the conception of the SASAR II project. A brief discussion of SAR theory is then given. The subject will not be dealt with in great detail here, as that is not the purpose of this document. The intention is merely to give an understanding of the basic SAR concepts and insight into why it has become so useful. It continues with a brief explanation of the entire SASAR II system. The aim of this section is to establish what role the transmitter and FDU play in the system. The concluding two sections of Chapter 2 provide the reader with the basic demands on the transmitter and FDU. They introduce the hardware from a functional standpoint and allow the topics of design and implementation to be tackled in the forthcoming chapters.

Chapter 3: Transmitter Design and Implementation

Transmitter Design and Implementation takes the high-level system of the previous chapter and, with the help of relevant theory, provides a practical way to implement hardware to realise it.

It begins with a theoretical discussion of the chirp generation process. A simulation is done to confirm the theory of this process. This is done using Systemview by Elanix. In essence, the first upconversion stage is outlined by this analysis. Some mixer and filter theory then explains their operation and the processes which contribute to unwanted signals appearing at the mixer output. Using the specified system frequencies, appropriate mixers and filters are selected. A system whereby one can ascertain the validity of each frequency stage is proposed.

The final transmitter section deals with the issue of power levels. This analysis will expose the need for amplification and finalises the transmitter design. The maximum power carrying capacity of the waveguide used is also established.

Chapter 4: Design and Implementation of the Frequency Distribution Unit

As with the previous chapter, we build here from the high-level ideas from Chapter 2. A means of generating the frequencies required from the FDU is presented. As with the transmitter, power analysis determines what amplification is required and lays out the FDU design. Unlike the transmitter though, many components of the FDU are fabricated rather than merely acquiring complete modules. Therefore, theory allowing one to understand the process necessary to generate these modules is presented. They are then fabricated and a final section deals with a means of analysing their performance.

Chapter 5: Testing

The penultimate chapter deals with testing of the system. The FDU is tested first as it is required in order to test the transmitter in entirety. The tests performed determine if the requirements on the modules that are generated have been met and continue by

drawing up a power budget for the FDU. This is followed by a similar power test on the transmitter. The final testing evaluates the performance of the complete system by analysing the signals that pass through it.

Chapter6: Conclusions and Recommendations

The final chapter concludes and presents recommendations for future work on the subject.

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Chapter 2

Background Information

2.1 Project Background

In early 2003, a South African consortium was established with the aim of consolidating and developing the local SAR capability. The main consortium members are Sunspace and Kentron. Both companies are interested in making use of SAR imagers in their respective products. For this reason, Sunspace launched the SAR Technology Project to achieve the ideas of the consortium and develop the following:

- * Simulations and evaluation of SAR processors
- * Applications studies
- * An X-band airborne demonstrator

The prime objectives of the airborne demonstrator project, which this dissertation forms part of, are stated by Sunspace to be:

1. To demonstrate a consolidated South African SAR ability to the local and international communities, by generating high quality images with a South African X-band demonstrator.
2. To identify and mitigate risks associated with a satellite payload, by first building an airborne SAR.

The initial choice of the airborne parameters, such as bandwidth and centre frequency, are aligned with technology that can be deployed on a micro-satellite SAR. These appear to follow a general trend toward high bandwidth, X-band systems [3].

2.2 Basic SAR Theory

Synthetic aperture radar (SAR) is an airborne radar mapping technique that generates high-resolution images of surface target areas and terrain. For this explanation of SAR, to be consistent with SASAR II case, we shall assume a conventional side-looking, pulse-compression, airborne radar. The setup is illustrated in Fig. 2.1 with the flight path into the page.

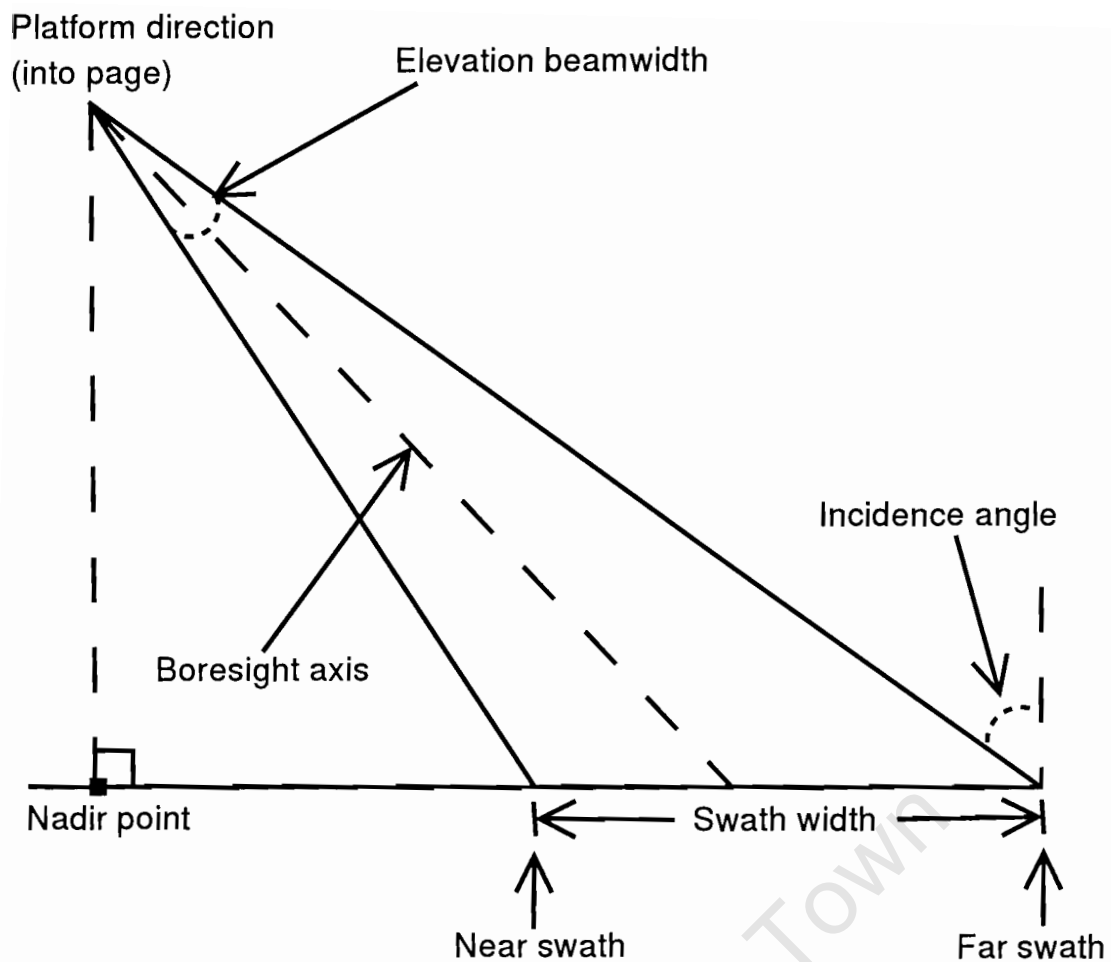


Figure 2.1: Illustration of a side-looking SAR setup viewed in the cross range direction.

The two most important terms to define are *slant range* and *cross range*. Slant range refers to the radar's line-of-sight range while cross range defines that transverse to slant range. Slant range resolution is commonly obtained by coding the transmitted pulse. This text will assume FM (Frequency Modulation) chirp coding. Resolution in the cross range is obtained by coherently integrating reflected energy as the radar travels above and alongside of the area to be mapped. The distance over which the radar platform collects reflectivity data and coherently integrates it in the cross range is termed the *synthetic aperture*.

As the radar moves alongside the mapping area, chirp pulses are emitted at some constant PRF. One of these pulses is shown in Fig. 2.2. The PRF is kept sufficiently low to ensure unambiguous range responses. For each transmitted pulse, the return signal is sampled at some range-sample spacing or recorded continuously over a pre-determined portion of the illuminated range extent (*range swath*).

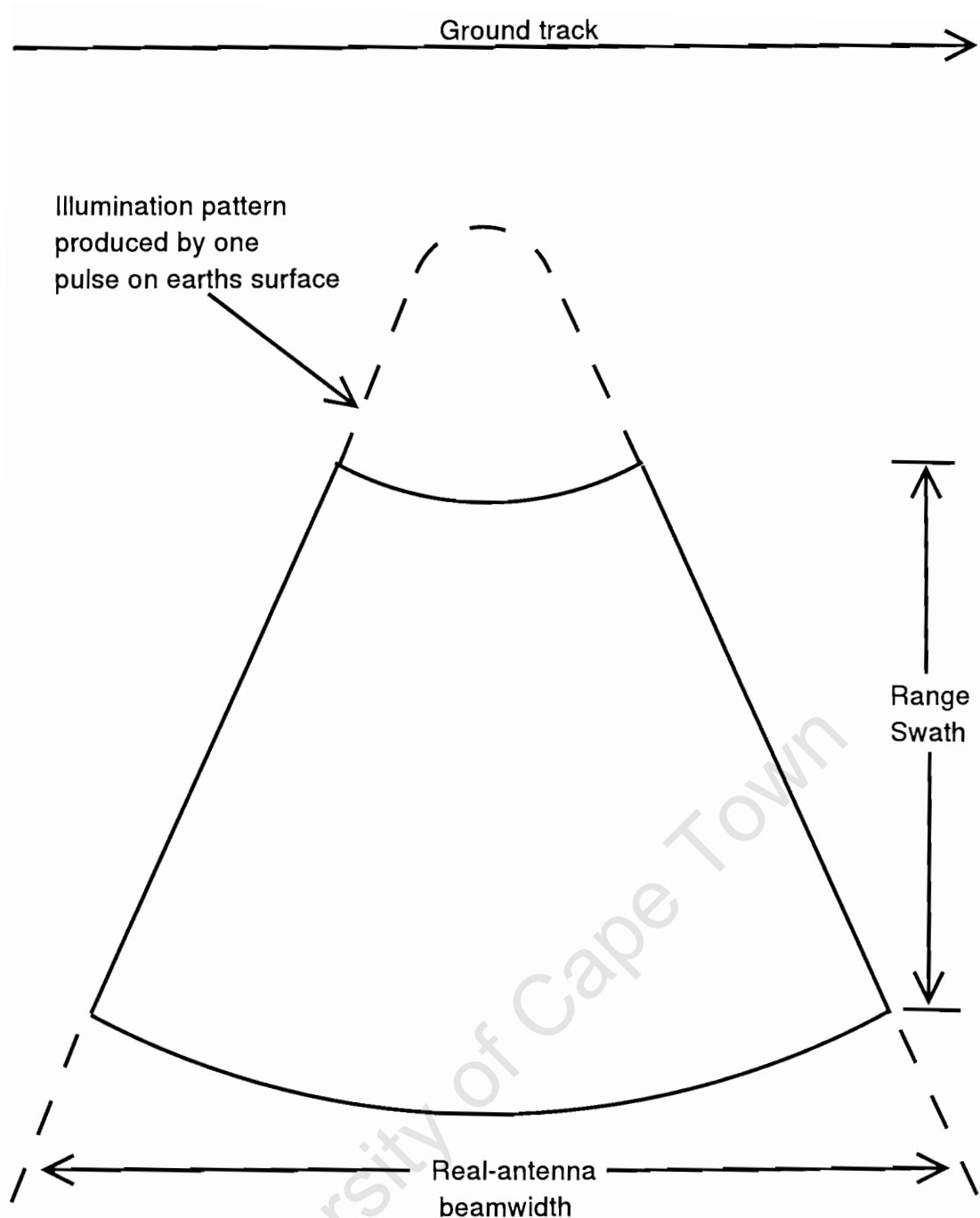


Figure 2.2: Illustration of the illumination pattern of one pulse on the earth's surface viewed from above.

The collected data is therefore a two-dimensional set of reflectivity measurements: slant range versus cross range. A data record will extend over the swath width in the slant range and along the flight path over which data is collected in the cross range. Data sampled from an echo signal of a single pulse is called a *range data line*, while that sampled in azimuth at a given range sample position is called an *azimuth data line*.

Prior to processing the data set will not resemble a map of the terrain. This is because echoes from isolated point targets are dispersed both in range and azimuth. In order for

this transformation to take place, range and azimuth compression must be performed on the data set. The reader is referred here to [2] for an explanation of this process as the topic is outside the scope of this document.

This technique has limitless applications which include target recognition, mapping and imaging functions. As well as the obvious military uses, applications extend to the fields of geology, agriculture and oceanography [2].

2.3 System Overview

The SASAR II project can be briefly described by the high-level system block diagram of Fig. 2.3. Each unit will be elaborated on in the forthcoming text in order to give the reader a greater insight into the project and establish the extent of this dissertation.

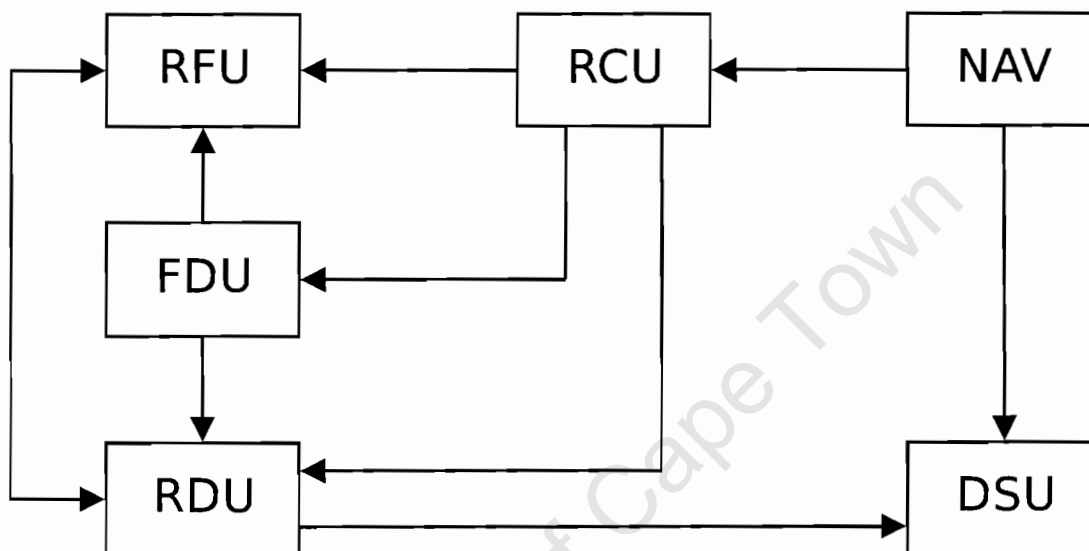


Figure 2.3: High-level system block diagram of SASAR II system.

The nine units that constitute this radar are:

1. The Radio Frequency Unit (RFU): This can be split into three sections, namely, the transmitter, receiver and antenna subassembly. It takes input from the RDU, RCU and FDU and outputs back to the RDU. A baseband chirp is received from the RDU. This is upconverted and amplified by the transmitter. The antenna subassembly, which consists of a duplexer and antenna, then radiates the incident power and collects back-scattered energy. This is passed to the receiver which performs amplification and down-conversion before returning the signal to the RDU.
2. The Frequency Distribution Unit (FDU): The FDU provides LO input to all mixers of the transmitter and receiver (transceiver). It also clocks the DAC and ADC to allow the RFU to 'communicate' with the RDU. The RCU initialises the output frequencies of the FDU.

3. The Radar Digital Unit (RDU): The RDU consists of the Digital Pulse Generator (DPG), Sampling Unit and Timing Unit. The DPG outputs the dual channel, base-band chirp to the RFU. The Sampling Unit then receives the IF response from the RFU and passes it on to the DSU. The Timing Unit sends trigger pulses to the DPG and Sampling Unit.
4. The Radar Controller Unit (RCU): Control of the RFU's testing system and TWT is given to this unit. Output frequencies of the FDU are also determined here. The final purpose of the RCU is to provide the the RDU with time information gathered from the Navigation Unit.
5. The Navigation Unit (NAV): This unit supplies the DSU with position and time information and the RCU with time information with the use of GPS (Global Positioning System) and a ring laser gyro.
6. The Data Storage Unit (DSU): This RAID server stores information from the NAV unit and RDU.
7. The Power Supply: This unit provides the necessary power to all the above units.
8. The Ground Segment: This is where all the post-processing is done on information stored in the DSU.
9. The Radar Platform: This aeroplane takes the form of a twin prop, pressurised Aero Commander 690 A to be hired from the South African Weather Services.

2.4 The Transmitter

The block diagram of Fig. 2.4 shows the five functional units of the transmitter. The dotted blocks indicate that the section is not part of the transmitter, but are included for clarity. The diagram was arrived at in accordance with the Terms of Reference (Section 1.1).

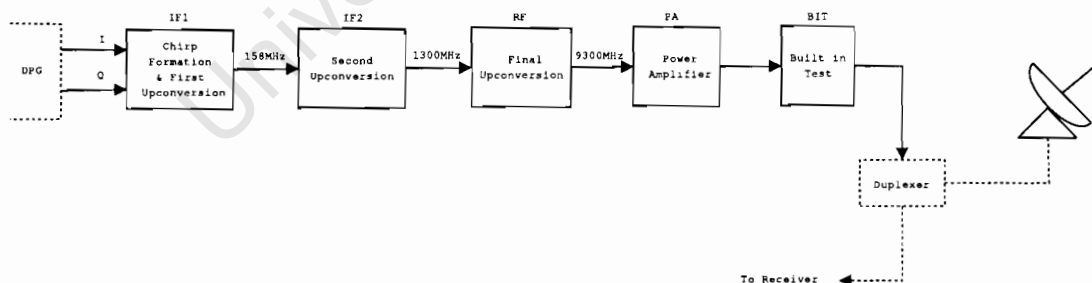


Figure 2.4: High-level block diagram of the transmitter.

The five functional sections are termed IF1, IF2, RF, PA and BIT. The transmitter interfaces with the DPG and the duplexer at its input and output respectively. All sections will be described briefly here in terms of their function.

2.4.1 Chirp Formation and First Upconversion (IF1)

The two, quadrature, baseband signals output by the DPG have a real bandwidth of 50 MHz. They need to be combined and upconverted to produce a chirp of 100 MHz bandwidth at the first system IF of 158 MHz. This will require a 158 MHz LO signal.

2.4.2 Second Upconversion (IF2)

A second IF is necessary in order to ease filtering parameters due to image frequencies. It has been chosen to be 1300 MHz, as stated in the Terms of Reference. This will make the system compatible with available Reutech L-band transmitter units, which facilitates transmitting at this frequency at a later stage. A LO of 1142 MHz will allow translation to this frequency.

2.4.3 Final Upconversion (RF)

The final upconversion sets the carrier frequency to 9300 MHz. This stage will be a functional repetition of the previous one, with a LO of 8000 MHz.

2.4.4 Power Amplifier (PA)

A *travelling wave tube* (TWT) performs the power amplification and passes the 3,5 kW chirp signal to the duplexer. The TWT is not dealt with in detail due to its unavailability during the time scale of this dissertation. Instead it is catered for by knowledge of its centre frequency and input and output power ratings. These are, as stated in the TWT's RF electrical specifications, 9.3 GHz, 14 ± 2 dBm and 3.2 - 4.56 kW respectively.

2.4.5 Built in test (BIT)

A simple testing function, whereby each of the above stages is verified, needs to be established. Therefore, if an error does arise, this makes it easier to pin-point the problem in the system. This will save time in locating the area of concern, and allow the operator to focus on rectifying the problem.

2.5 Frequency Distribution Unit

The FDU has, as for the transmitter, been broken up into functional blocks. This unit needs to generate high quality clock signals for the DPGs DAC and the ADC. LO signals for all mixers of the transmitter and receiver (transceiver) also require generation. In order to maintain coherency, all frequencies generated by this unit must be locked to a stable reference. Frequency synthesizers can provide this service, with a 10 MHz GPS crystal as the reference. Of course, the 10 MHz needs to be split in order to be delivered to each synthesizer. All required signals are briefly explained now.

2.5.1 DAC and ADC Clocks

A 150 MHz, 1.0 V_{p-p} signal needs to be generated with a maximum jitter specification of 6 ps as per the Terms of Reference (Section 1.1). This requires good phase noise characteristics, the frequency domain equivalent of jitter (the time domain version). The 210 MHz clock should be identical to the 150 MHz clock in terms of function and design.

2.5.2 Mixer Drives

Three frequencies need generation for LO signals. As stated before, they are 158 MHz, 1142 MHz and 8000 MHz. The receiver requires only the latter two, as sampling is performed at the first IF.

Chapter 3

Transmitter Design and Implementation

3.1 Chirp Theory and Simulation

The DPG outputs two quadrature, analogue channels (I and Q), where $I = \cos(\pi Kt^2)$ and $Q = \sin(\pi Kt^2)$ (both real, even signals). K is known as the chirp rate ($K = B/T$ in Hz/s) [4]. The two chirp pulses vary linearly in frequency from -50 to 50 MHz in a period of $5 \mu\text{s}$ (T). It can be noted that the concept of a negative frequency here can be resolved as follows:

$$\begin{aligned}\cos(-\pi Kt^2) &= \cos(\pi Kt^2) \\ \sin(-\pi Kt^2) &= -\sin(\pi Kt^2)\end{aligned}$$

The I and Q channel graphs are plotted below (Fig. 3.1 and 3.2) as output by the DPG (ideally). Their analogue bandwidths are 50 MHz.

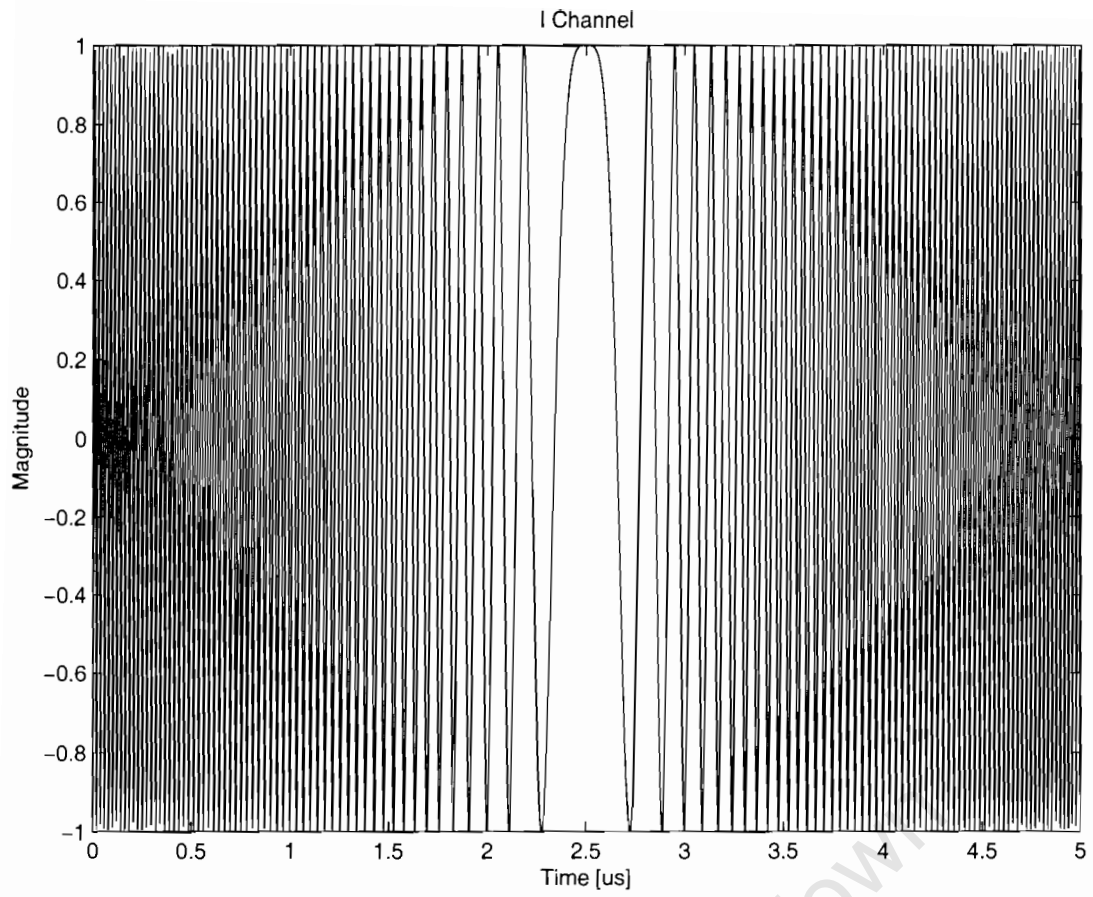


Figure 3.1: I channel output of DPG (Matlab simulation)

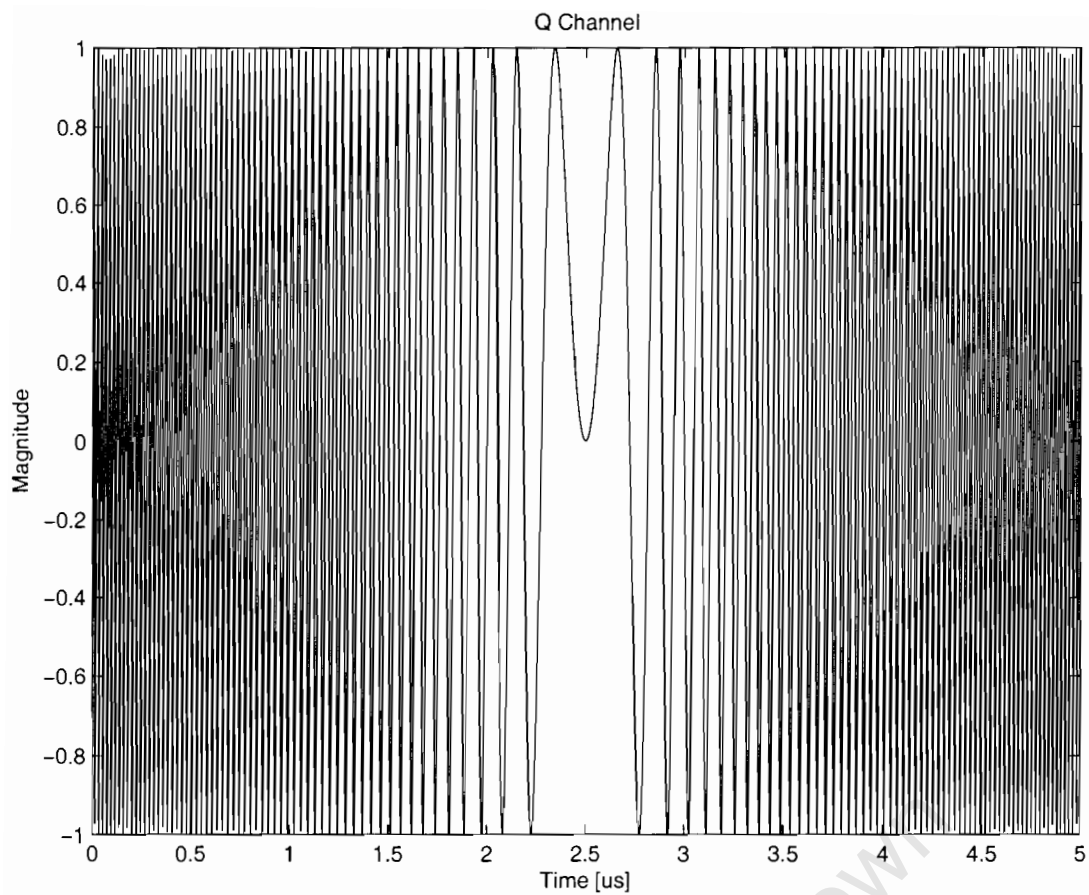


Figure 3.2: Q channel output of DPG (Matlab simulation)

These two signals are respectively multiplied with a 158 MHz sine or cosine waveform (irrespective of order). This is equivalent to upconversion with a common LO with one arm being shifted by 90 degrees. Figs 3.3 and 3.4 are the result of these two operations.

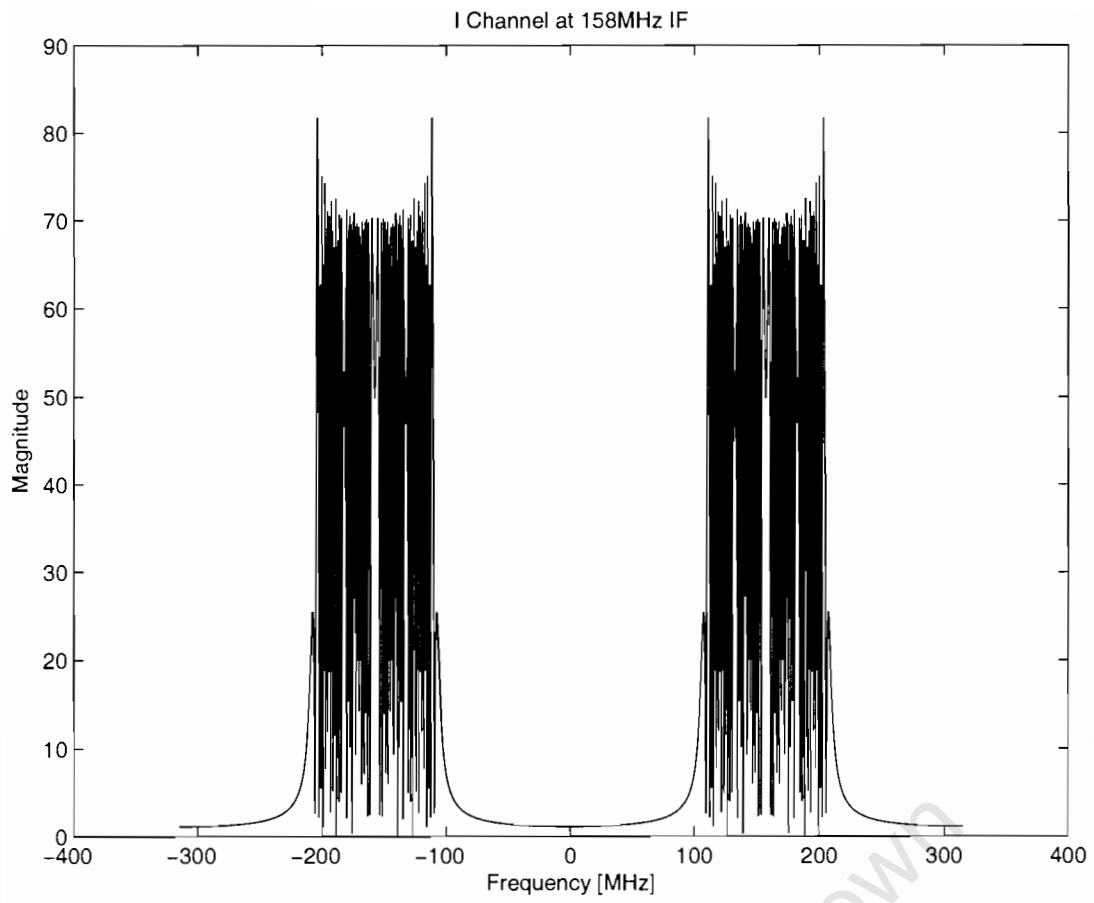


Figure 3.3: I channel spectrum at 1st IF (Matlab simulation)

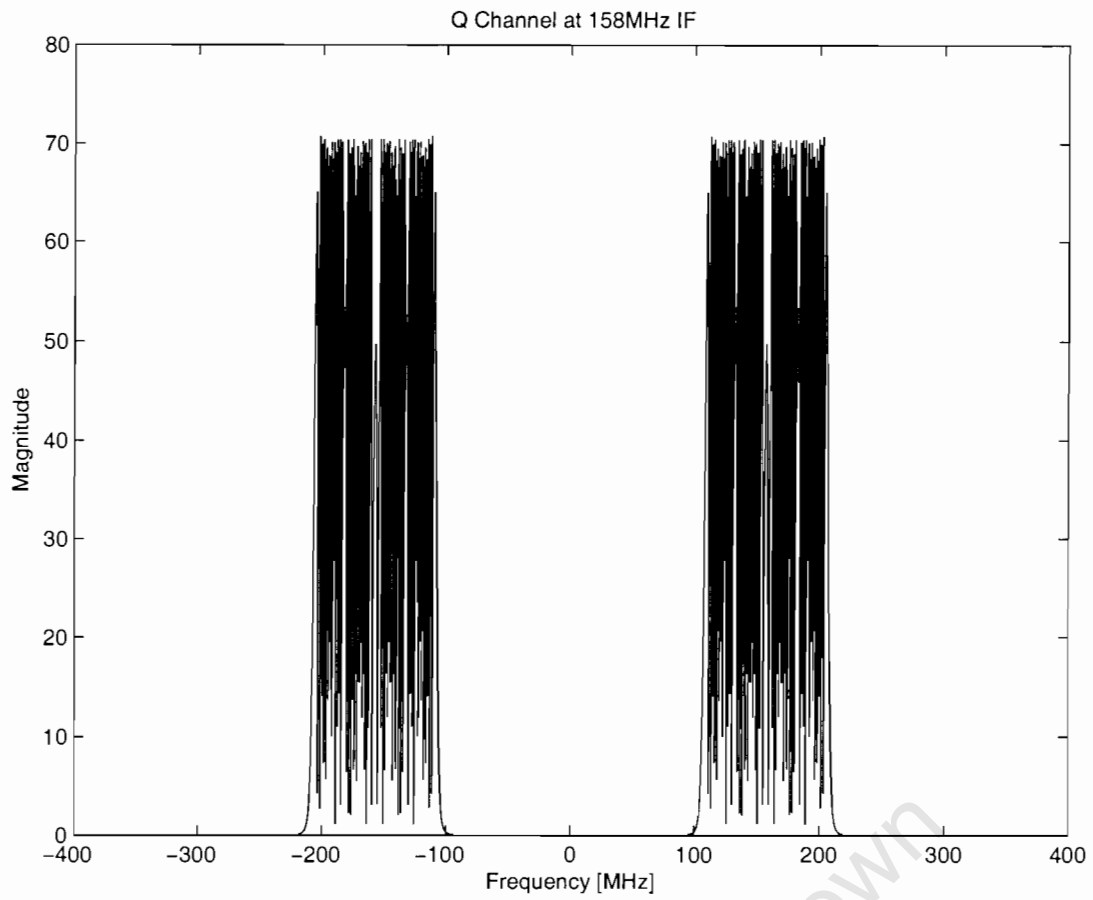


Figure 3.4: Q channel spectrum at 1st IF (Matlab simulation)

They are then combined at this frequency to produce the complex signal: $Chirp = (I + jQ)_{IF}$. This has a bandwidth of 100 MHz. The Matlab code for these graphs is included in Appendix A [7].

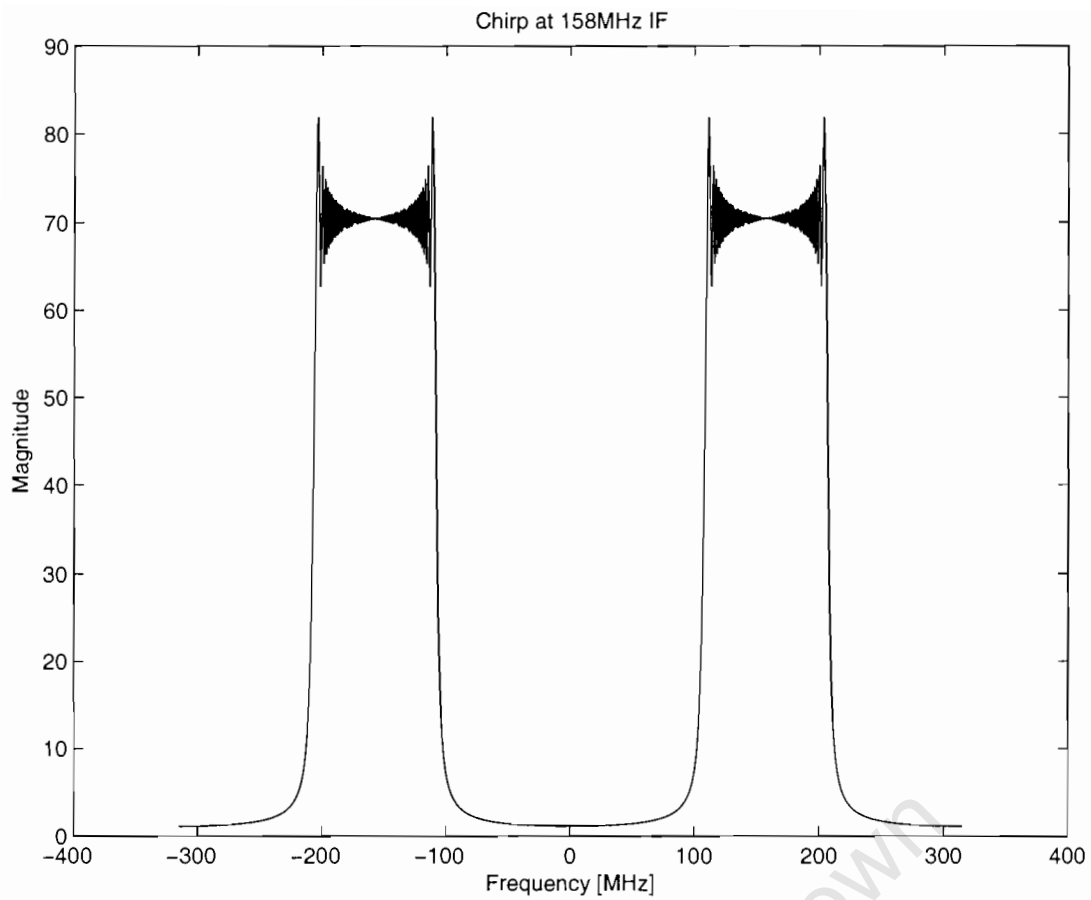


Figure 3.5: Chirp spectrum at 1st IF (Matlab simulation)

The setup, which appears in Fig. 3.6, simulates the above process with a time-domain processor called Systemview (by Elanix). The two baseband channels are shifted up in frequency by LO signals differing in phase by 90 degrees with identical mixers. These two 158 MHz signals are then combined to produce the chirp signal. A 5th order Butterworth filter is applied to the output.

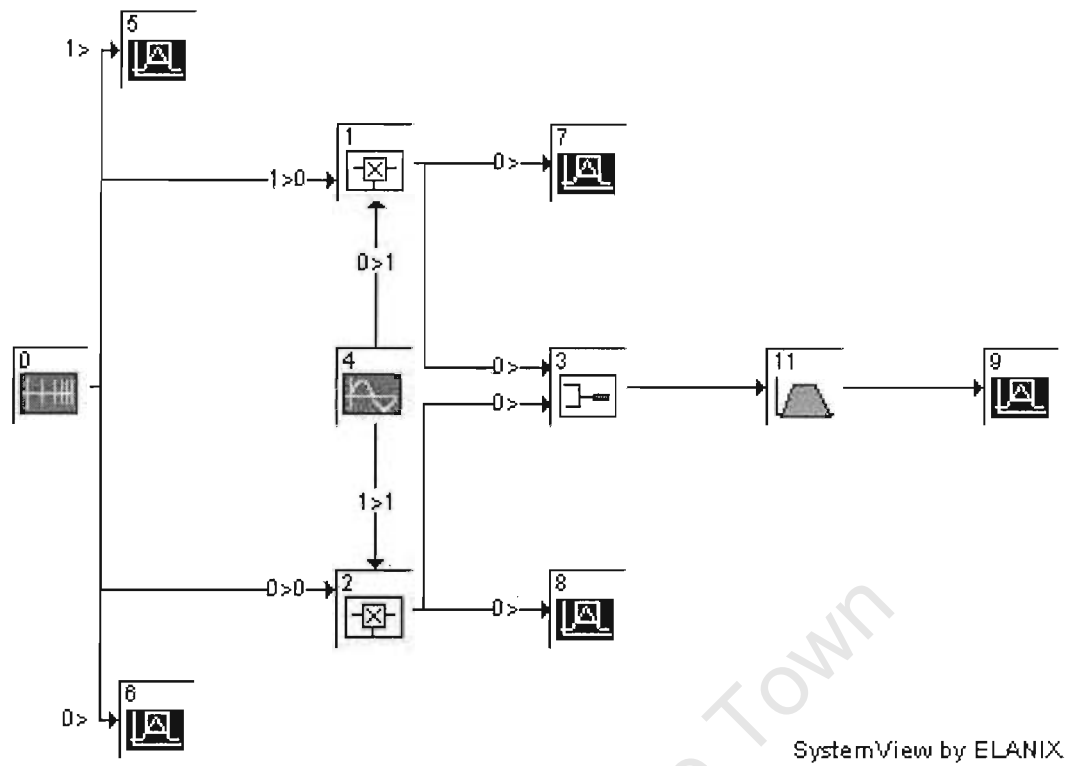


Figure 3.6: Systemview setup for simulation of IF1.

The simulation results follow:

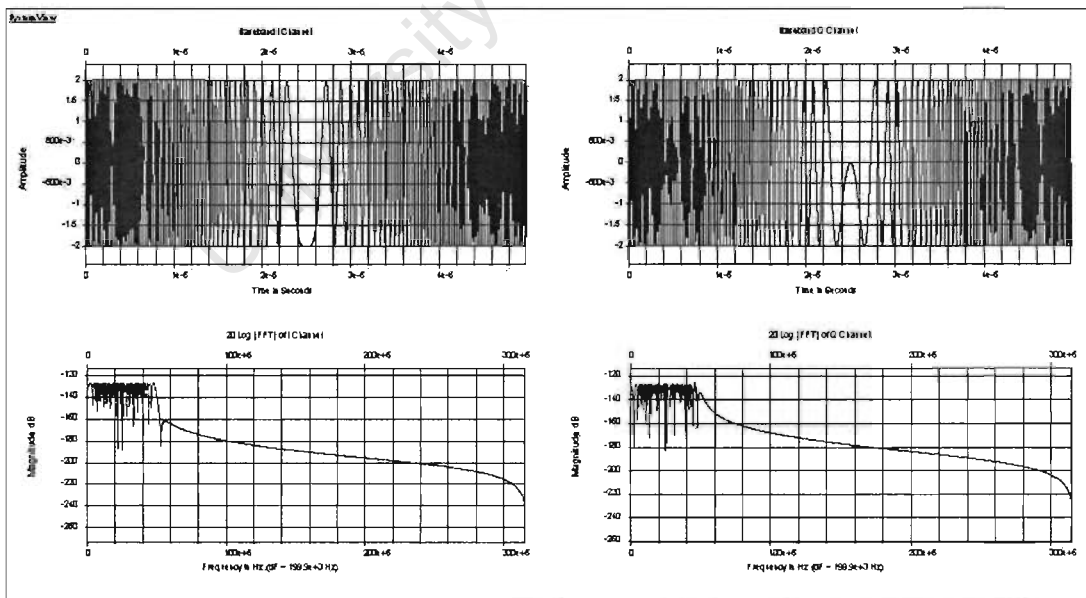


Figure 3.7: Time and frequency domain graphs of in-phase and quadrature chirp signals at baseband.

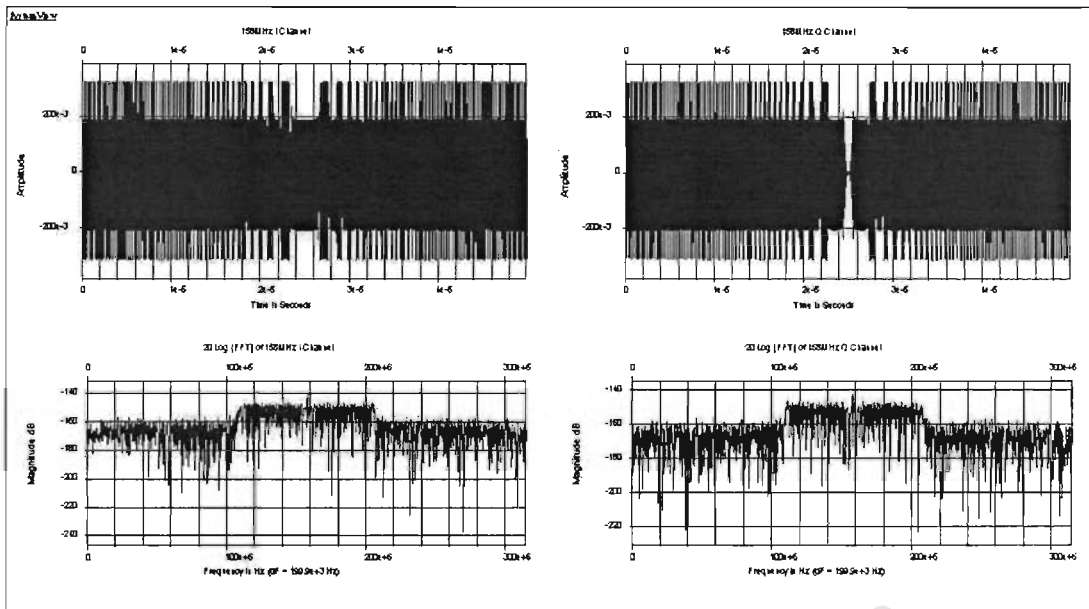


Figure 3.8: Time and frequency domain graphs of in-phase and quadrature chirp signals at 158 MHz (IF1).

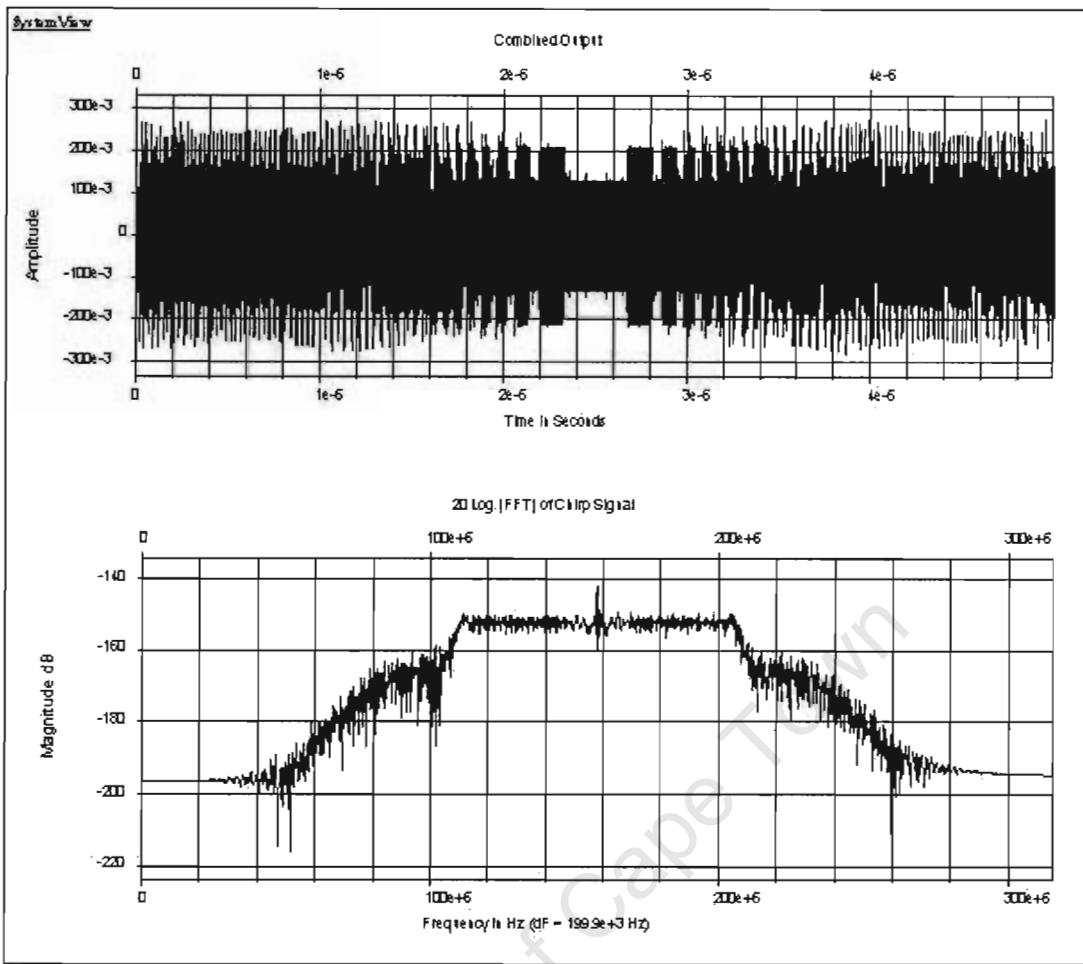


Figure 3.9: Resultant complex chirp signal.

3.2 Mixers and Filters

Mixers are used to translate a signal in frequency (either up or down). The RF signal to be translated is combined with a LO signal in a non-linear element such as a diode. It outputs $nLO \pm mRF$, where m, n are integers or zero. These sums and differences are referred to as the mixers harmonic intermodulation products. Common types are the unbalanced, single balanced, double balanced and image reject mixers. The important factors to consider are minimum LO and maximum RF input power level, conversion loss, port-to-port isolation, return loss and of course frequency.

For up-conversion in the SASAR2 transmitter, the upper sideband (LO+RF) is the only desired output and all other signals need to be filtered out. The output of each of the transmitter sections needs to be ascertained to ensure unwanted signals are attenuated sufficiently before entering the successive stage. Essentially, this analysis details the required filter characteristics for each of the three up-conversion stages.

The three properties: bandwidth, steepness of the transition from passband to stopband (roll-off) and delay time are of great interest in selecting a bandpass filter in a wideband radar system. The degree at which a filter of defined bandwidth 'rolls off' is determined by the number of poles it incorporates. However, delay time increases with an increase in the number of poles and decreases with an increasing bandwidth. Therefore, a compromise must be made. It is important to make the filter bandwidth reasonably larger than the signal bandwidth as can be seen from Fig. 3.10. If this is not done, delay at the outer frequency components will differ from those at the central frequency components of the spectrum. This phenomenon is known as group delay and causes the pulse envelope to vary from input to output [2, 9]. This is undesirable as it compromises the resolution of the SAR system. The user-requirements state that Butterworth filters are to be used. This is because they incorporate the best trade-off between flatness in the passband, degree of roll-off and flatness of time delay [11]. Since 5th order filters are to be used, the only parameter that may be varied is the bandwidth.

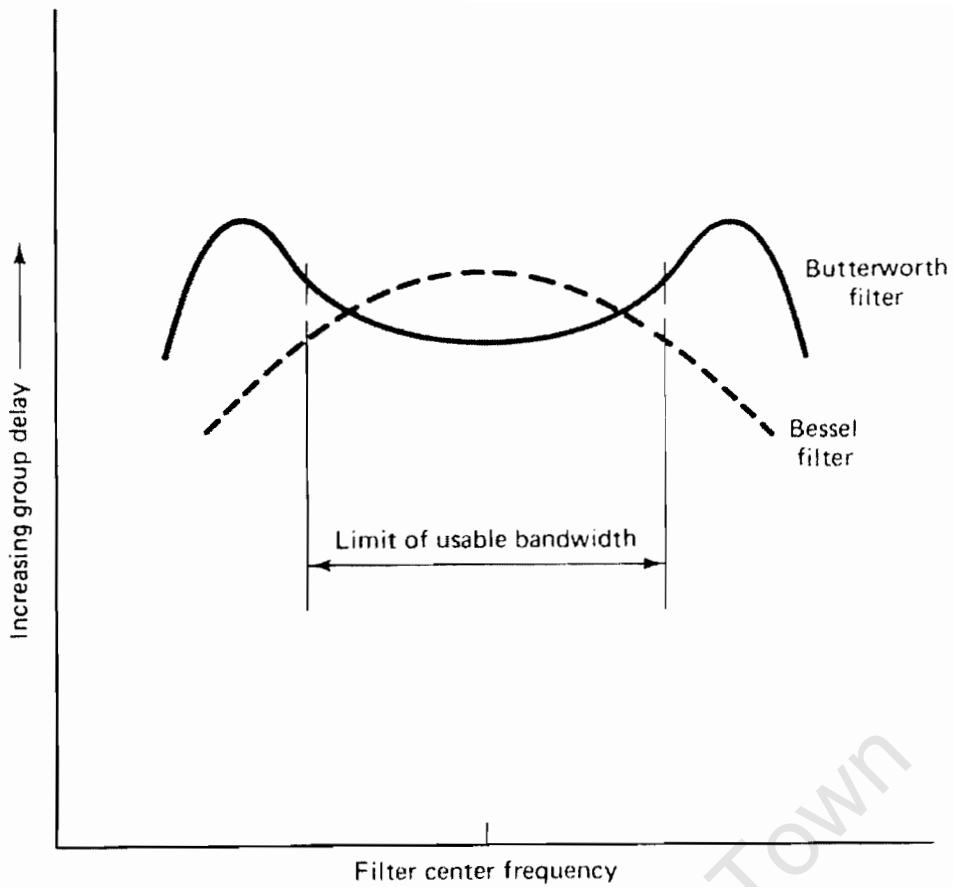


Figure 3.10: Group delay near a filter's centre frequency [9].

With the above material taken into consideration, we can now move on and detail the required mixers for each of the up-converter stages.

3.2.1 IF1

Since the first upconversion stage takes a baseband input, all outputs are centred on LO harmonics. This is explained by Table 3.1.

Table 3.1: IF1 mixer harmonics with LO=158 MHz and RF=0 MHz.

n	m	nLO+mRF	nLO-mRF	Order
0	0,1,2,3...	0	0	1
1	0,1,2,3...	158	158	2
2	0,1,2,3...	316	316	3

Fig. 3.11 illustrates what is stated in Table 3.1. Superimposed on the diagram is the harmonic intermodulation data for a Mini-circuits double-balanced mixer (ZFM3). An

ideal brick-wall bandpass filter centred at 158 MHz could have a maximum theoretical passband of 216 MHz in order to ensure only LO+RF is presented to the next mixer. However, for a realistic filter, it is possible that signal feed-through (RF) and second harmonic (2LO+RF) will not be attenuated below the noise floor. The darker dashed line shows the 5th order Butterworth response, while the dotted line displays its effect on the spectrum. The effects of these adjacent signals entering the mixer of IF2 must therefore be considered.

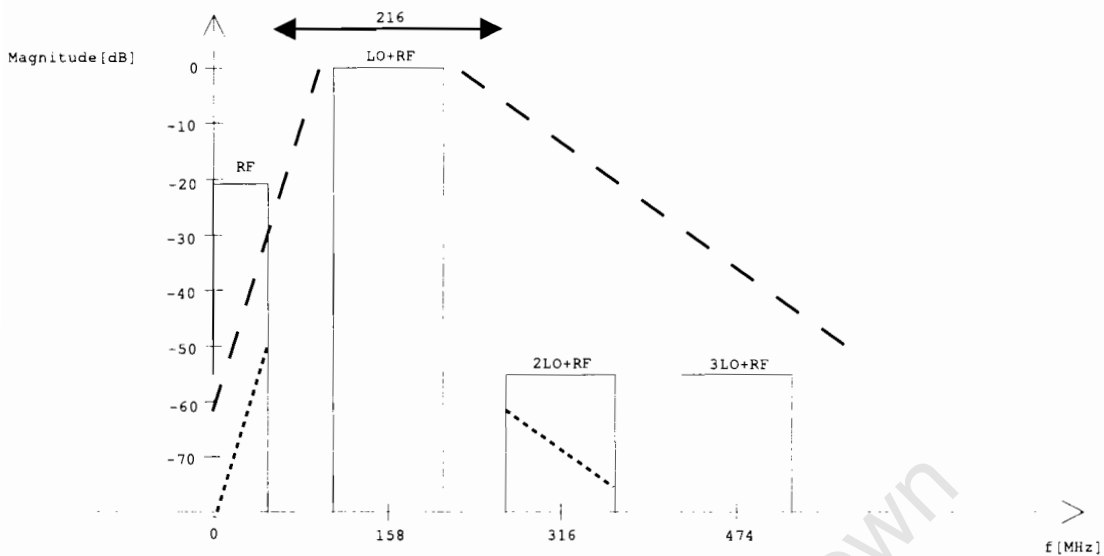


Figure 3.11: IF1 harmonic intermodulation (relative to desired IF output) for Mini-circuits ZFM3 mixer with filter attenuation.

3.2.2 IF2

Signal feed-through from the IF1 mixers would be translated by this mixer to be centred at 1142 MHz (LO). The other signal that would possibly enter this mixer (2LO+RF) would sit where this mixers LO+2RF sits (see Table 3.2). This is because the first stage was mixing up from baseband and therefore, the LO frequency of the first stage is equal to the RF input of the second stage. So, all harmonics of the first stage will mix up to sit exactly where harmonics of the second stage sit. Therefore, the filter requirements for the output of IF2 should be more stringent than for IF1. If the second filter is to remove its harmonics, then the previous stages harmonics will also be removed here. This allows the requirements of the first filter be relaxed.

Table 3.2: IF2 mixer output with LO=1142 MHz and RF=158 MHz.

n	m	nLO+mRF	nLO-mRF	Order
0	1	158	-158	1
1	0	1142	1142	1
1	1	1300	984	2
1	2	1458	826	3
1	3	1616	668	4
2	6	3232	1136	8
2	7	3390	978	9

Once again, mixer harmonic intermodulation data and the effect of filter attenuation is shown in Fig. 3.12 for a Mini-circuits ZLW-5 double-balanced mixer. The image and LO feed-through are found to be sufficiently large to warrant alarm bells.

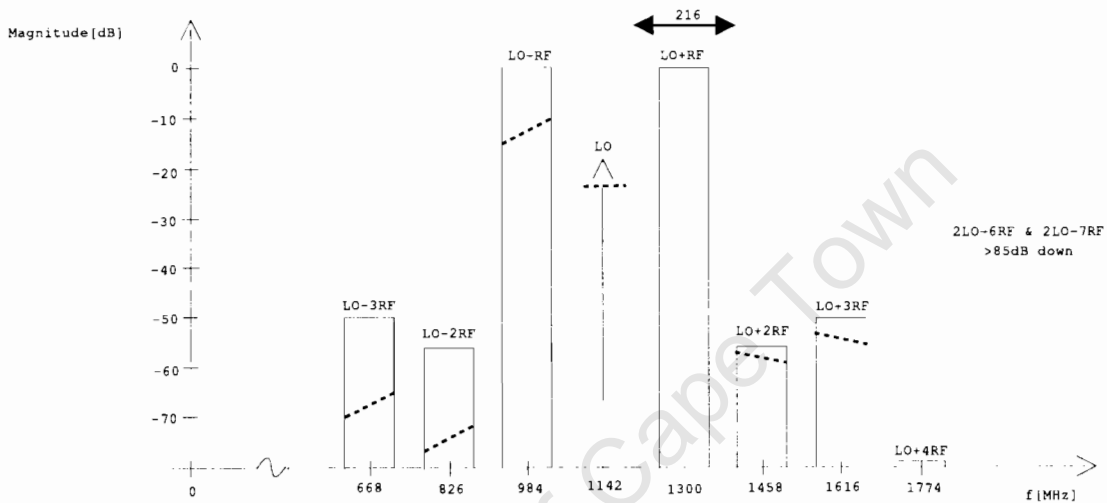


Figure 3.12: IF2 harmonic intermodulation (relative to desired IF output) for Mini-circuits ZLW-5 mixer.

This problem can be vastly improved by replacing the double-balanced mixer with an image rejection mixer. Therefore, a Miteq IR0102LC1C mixer is used here. This mixer boasts image rejection of 23 dB and LO-RF isolation of 40 dB.

3.2.3 RF

Analysing this mixer, without giving attention to harmonics from the previous, gives rise to Table 3.3 and Fig. 3.13.

Table 3.3: RF mixer output with LO=8000 MHz and RF=1300 MHz.

n	m	nLO+mRF	nLO-mRF	Order
0	1	1300	-1300	1
0	6	9100	-9100	6
0	7	10400	-10400	7
1	0	8000	8000	1
1	1	9300	6700	2
1	2	10600	5400	3
2	5	22500	9500	7
2	6	23800	8200	8

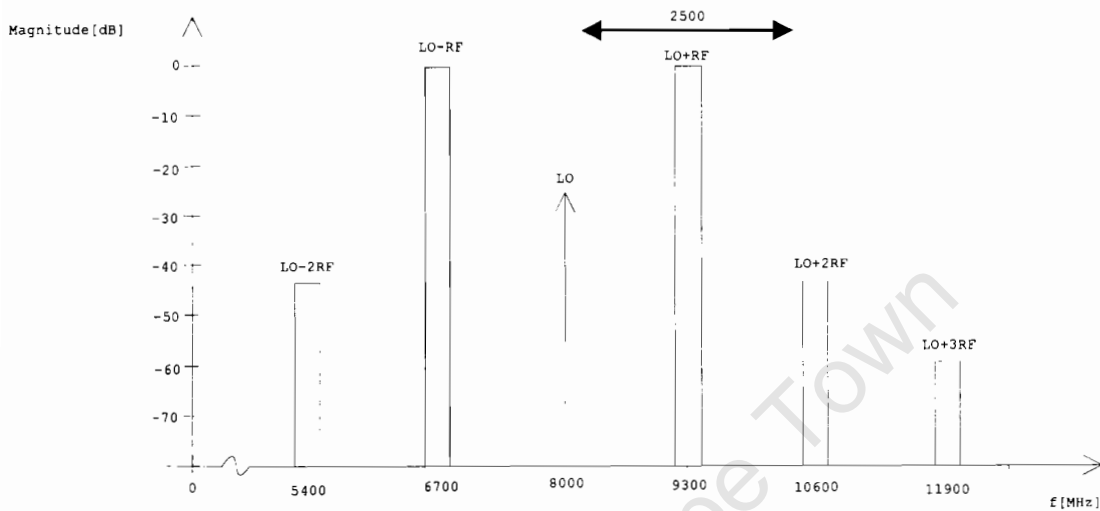


Figure 3.13: Third mixer harmonic intermodulation (relative to desired IF output). Miteq MO812M5

This stage has far less critical filtering requirements than the previous two. However, if harmonics on either side of LO+RF from the previous stage are considered to be let through, then the filters passband for the final mixer is reduced to that of the previous two stages. This makes the role of the IF2 filter even more important. For this reason and because the TWT has a narrow input frequency range, no filter is included here.

3.3 BIT

The output of each frequency stage is diverted through a *single pole double throw* (SPDT) switch to the equivalent section in the receiver. This is achieved by applying a TTL high (+5 V) signal to the control input of the switch. The RCU shall perform this function. Therefore, with no signal applied to the switches, the transmitter will perform its normal function. Table 3.4 shows the states of this function. This verifies the design up to the power amplifier input.

Table 3.4: Transmitter states with associated switch control signals.

IF1 switch	IF2 switch	RF switch	Transmitter State
0	0	0	Transmit mode
0	0	1	RF verification
0	1	0	IF2 verification
0	1	1	-
1	0	0	IF1 verification
1	0	1	-
1	1	0	-
1	1	1	-

A test function is now developed to confirm the operation of the TWT. This consists of a directional coupler, transfer switch, dummy load and two power detectors (see Fig. 3.14). The transfer switch with dummy load allow us to access full TWT power without the inherent dangers of transmitting at this high power. This is purely a testing facility. The directional coupler has power detectors on both coupled arms to monitor forward and reverse power. The transfer switch will be switched to dummy load to ensure correct power levels and then back to transmit before transmission. It will then remain here until power down. After peak level detection, the analogue values are sent to the RCU for digitisation and interpretation. Power levels are still monitored during transmission, however, and if boundary conditions are exceeded the TWT can be shut down.

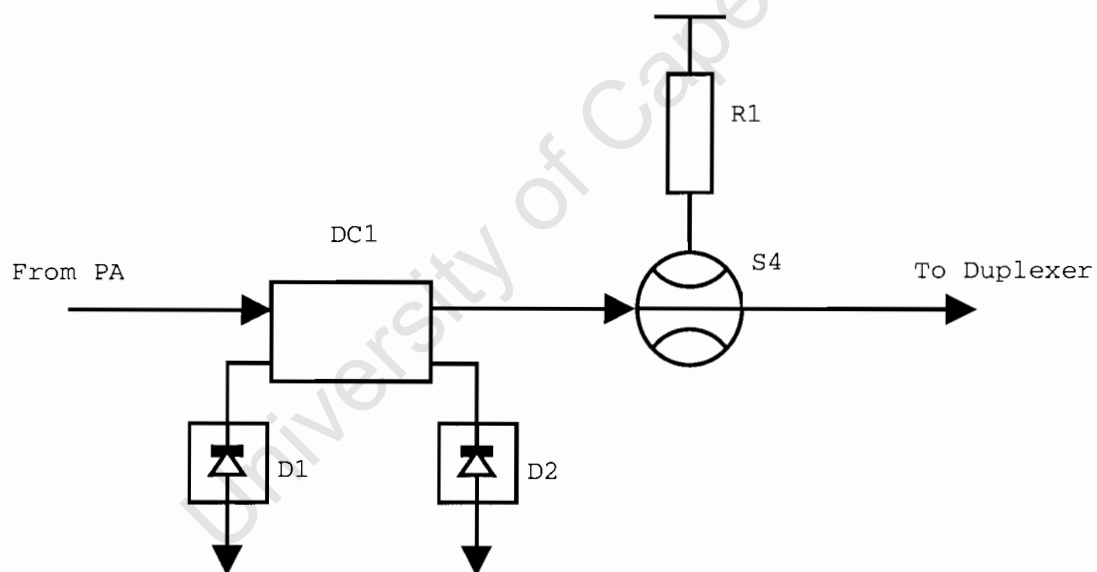


Figure 3.14: Diagram of the BIT.



Figure 3.15: Photo of the waveguide BIT components attached to the duplexer.

3.4 Power Requirements of Proposed Transmitter

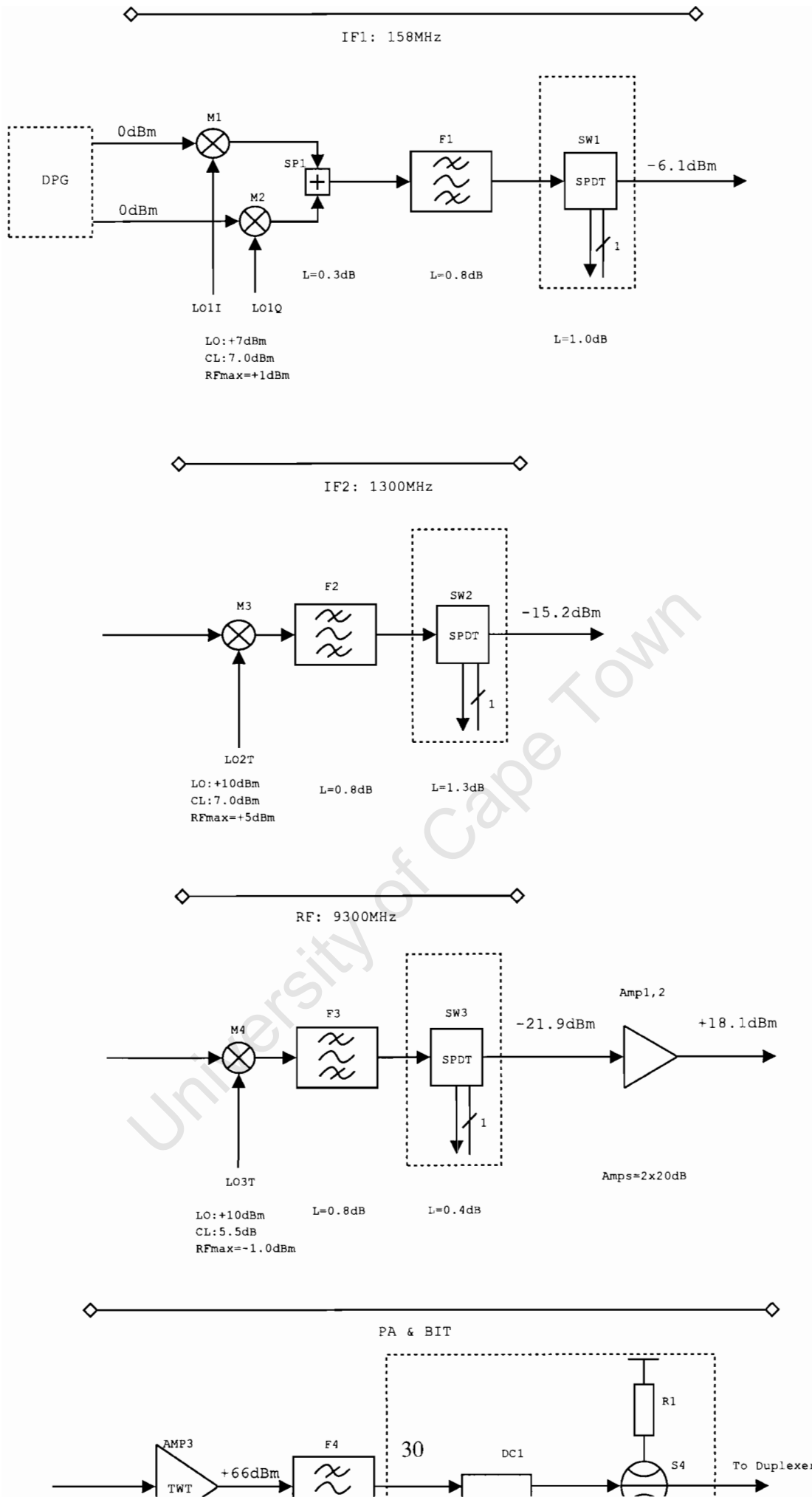
With a viable transmitter now realised, the power levels need to be tracked through the system to ensure all components meet their input specifications. This analysis will determine where amplification/attenuation is required. Since the TWT output requires wave-

wide transmission lines, analysis also needs to be done to determine if they are capable of handling the high power.

3.4.1 Power Levels

The Terms of Reference state that the DPG outputs two +10 dBm signals. This input level requires a mixer with a very high LO power level. This will put pressure on the FDU to attain this power specification. Therefore, these signals are attenuated in order to deliver 0 dBm to each IF1 mixer. The proposed system is shown in Fig. 3.16. All mixer specifications are stated below the component in the diagram. Losses for other components are also shown. This diagram includes BIT components in the dotted boxes. All connections are 50 Ω , SMA female.

University of Cape Town



It must be realized though, that until components are put on the test bench and power levels measured, values obtained from data sheets are merely estimates. They are the estimated lowest values ie. worst case. Also, losses due to connecting cables have not been taken into account. It is for this reason that, for all components where input power is crucial, levels are overstated by at least 3-4 dB. This will ensure correct operation with the use of barrel attenuators to pad levels down to specific numbers. The use of attenuators also acts to reduce the effects of reflections in the system, which are padded down by twice the attenuation value for a two way reflection path.

With this noted, and the fact that the TWT requires a +10 dBm input, two 20 dB amplifiers are inserted at the RF output. This raises the power level from -21.0 dBm to +18.1 dBm.

3.4.2 Waveguide Power-carrying Capacity

The maximum power which may pass through a waveguide will depend on the maximum electric field strength that can exist without breakdown . Experimental data on allowable field strengths at ultra-high frequencies indicates a value of 30 000 volts/cm applicable for air-filled guides under standard sea-level pressure, temperature and humidity conditions. With this maximum allowable field strength specified, the maximum power-carrying capacity of the waveguide can be specified as:

$$P = E_{max}^2 6.63 * 10^{-4} ab \left(\frac{\lambda}{\lambda_g} \right)$$

where a and b are in cm and P is in watts. Here a and b are the dimensions of a rectangular waveguide operating in the TE₁₀ mode with $a > b$ [12].

With $\lambda = 3.226$ cm, $\lambda_g = 4.175$ cm, $a = 2.54$ cm and $b = 1.27$ cm, we get:

$$P = 1.487 \text{ MW}$$

This figure assumes sea-level conditions. However, with increasing altitude it is found that the potential gradient at which breakdown occurs decreases because of the change in atmospheric pressure. Taking this into account and assuming that Paschen's law holds for ultra-high frequencies and assumes a seasonal average air density, it can be seen that at 10 000 ft (platform height), the maximum power must be derated by fifty percent. This gives a theoretical maximum power-carrying capacity:

$$P = 743.5 \text{ kW}$$

Therefore, depending on pressurisation of the aircraft, the figure should sit somewhere between these two values. This far exceeds the power rating of the TWT.

3.5 Summary

The theory of combining an in-phase and quadrature chirp signal was introduced. Through simulation of this, the hardware of the first IF stage was initialised. Some mixer and filter

theory then lead into analysis of specified mixer harmonic intermodulation rejection. The first filter was not deemed critical and its bandwidth is set to 140 MHz. The second filters' role was determined to be crucial and therefore its bandwidth is set to 120 MHz and the double-balanced mixer was exchanged for an image rejection mixer. The third filter had, potentially, the least critical constraints. Because of the TWTs narrow input frequency range it was thus excluded. A testing function was then outlined which sequentially monitors the output of each frequency stage.

With the transmitter all but designed, a power budget was drawn up which introduced the need for amplification. Analysis was done on the power-handling capability of the waveguide used. It was found that power levels do not pose threat to the breakdown of air. Table 3.5 summarises the components used in the transmitter. Fig. 3.16 can be referred to for component name designation.

Table 3.5: Components selected for the transmitter.

Component name	Manufacturer	Part number
AMP1	Tellumat	Custom
AMP2	Tellumat	Custom
AMP3	-	-
D1	Advanced Control	ASCP2504PC3
D2	Advanced Control	ASCP2504PC3
DC1	Hewlett Packard	X750E
FL1	Lorch	Custom
FL2	Lorch	Custom
FL3	-	-
FL4	-	-
FL5	Lorch	Custom
MIX1	Mini-circuits	ZFM3
MIX2	Mini-circuits	ZFM3
MIX3	Miteq	IR0102LC1C
MIX4	Miteq	MO812M5
R1	Sperry	566
S1	Sivers Lab	730 5X
SP1	Mini-circuits	ZMSCQ-2-180
SW1	Mini-circuits	ZYSW-2-50R
SW2	Mini-circuits	ZYSW-2-50R
SW3	Mini-circuits	MSP2T-18

Chapter 4

Design and Implementation of Frequency Distribution Unit

With the transmitter designed, it is now necessary to move to the design of the FDU, which will provide drive to the transmitter. As discussed in Section 2.5, a 10 MHz reference is split and delivered to the various synthesizers. With this in mind, the synthesizers are discussed and selected. The issue of power levels is then addressed with the intention of specifying required amplification. Some theory follows which will enable the synthesizer and amplifier hardware to be realised. A parameter whereby the performance of the FDU outputs is measured is then discussed.

4.1 Frequency Synthesizers

A frequency synthesizer is a subsystem that can derive a large number of discrete frequencies from an accurate, highly stable crystal oscillator. The reference crystal source determines the frequency stability and accuracy of the derived frequency. Fig. 4.1 shows how a frequency synthesizer can be realised using a phase locked loop (PLL) and a programmable frequency divider (N). The addition of a frequency divider (R) at the reference input will improve resolution. For the phase detector output to be zero the following condition must be met:

$$\frac{f_O}{N} = \frac{f_R}{R}$$

Therefore

$$f_O = \frac{N}{R} f_R$$

A number of frequencies may be attained by varying $\frac{N}{R}$, the division ratio [15, 2, 9].

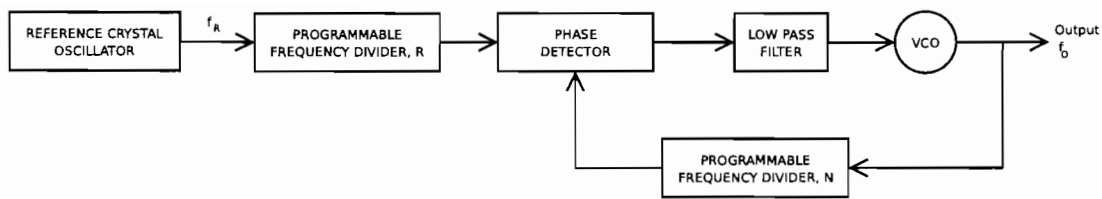


Figure 4.1: Block diagram of a frequency synthesizer using a PLL and programmable frequency synthesizers.

With the theory clear, the applicable synthesizers are chosen. They are the Synergy SPLL-A40 (150-250 MHz), SPLH-S-A79 (1040-1185 MHz) and SPLH-4000F (4000 MHz). Fig. 4.2 shows the configuration for the SPLH-S-A79. Only this particular synthesizer will be explained owing to its similarity to the SPLL-A40. The SPLH-4000F is a fixed frequency device and therefore does not require programming.

The external reference oscillator is divided down in the 14 bit reference divider and compared to the VCO frequency which is divided by the 18 bit N counter. The error signal, which is proportional to the phase difference between the two signals, is fed to a loop filter. This filter has been designed for maximum reference sideband rejection and optimum acquisition speed.

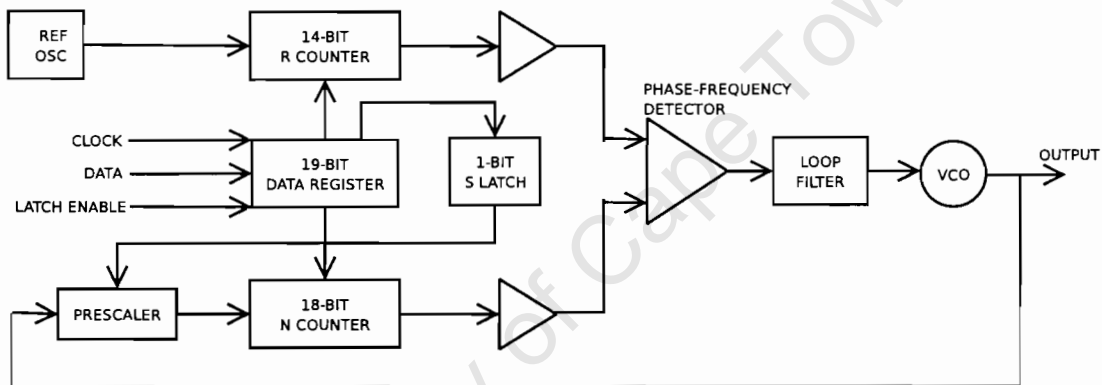


Figure 4.2: Block diagram of the Synergy SPLH-S-A79 frequency synthesizer.

The synthesizer takes serial inputs on the Clock, Data and Latch Enable pins to program the two counter dividers. The Clock input latches in a 19 bit word on the Data pin on the rising edge of each pulse. If the Control bit (last bit input) is high, then data is transferred to the R-counter and S Latch (prescaler select). If low, data is transferred to the N-counter. The Latch Enable input goes high after each Control bit is input. National Semiconductors Codeloader2 program [14] can be used to calculate the required outputs and program the synthesizer. This is done via the parallel port of a computer.

The 4000 MHz output of the SPLH-4000F is doubled to achieve the required 8000 MHz LO drive. This is done using the Mica F15KSP frequency doubler.

4.2 Power Issues

Once again, the power levels need examination to ensure outputs meet their requirement. Fig. 4.3 illustrates the proposed FDU design, with necessary amplifiers included. As with the transmitter, all connections are $50\ \Omega$, SMA female.

The 10 MHz reference outputs a 0 dBm signal. This is input to a Mini-circuits ZBSC-615 six way splitter which has a loss of -8.3 dBm. This poses the problem of not providing sufficient input power for the frequency synthesizers (+6.0 dBm). Therefore, an amplifier is required to boost the power by at least 14.3 dB. This prompts the insertion of five 16 dB amplifiers. The extra gain is accounted for by reasons stated in Section 3.4.1. By placing the amplifiers on the splitter output, instead of one at the input, we achieve greater isolation. This process is carried to the remaining three splitters.

The calculated power level is stated at the final outputs of Fig. 4.3. This is accompanied by the minimum required power. All components have also been named for purposes of clarity. A $50\ \Omega$ termination is placed on the sixth output of SP3.

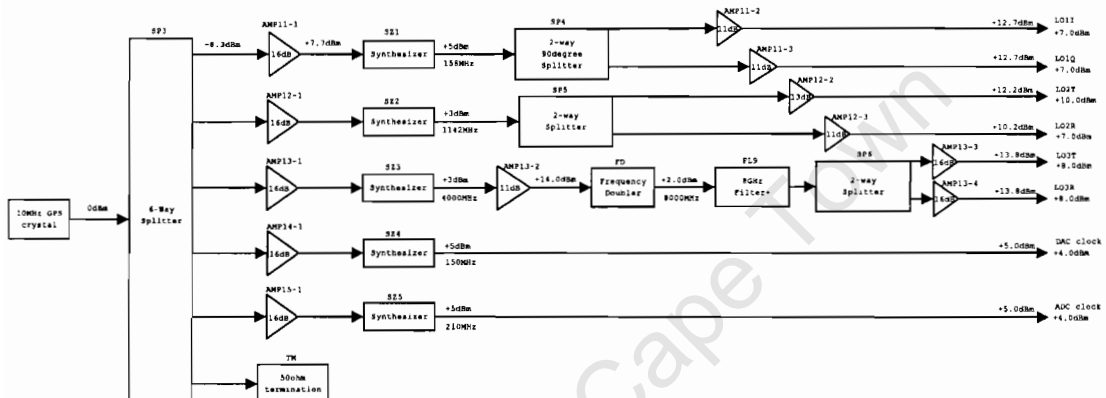


Figure 4.3: Diagram showing the proposed FDU design with predicted power levels.

4.3 Circuit Board Design

With the components of the FDU now set, the synthesizer chips and surface mount amplifiers need to be placed on printed circuit board (PCB). To be consistent, input/output impedances need to be maintained at $50\ \Omega$.

A transmission medium is required in order to transmit energy from one point to another. Some kind of structure is needed if the medium is not air (in the case of RF/microwave communication links). Traditionally, coaxial or waveguide technology has been used to connect together prefabricated circuits and components in the broad microwave range from 0.1-30 GHz. More recently, microwave integrated circuits (MICs) have been developed where components are connected and manufactured using the same technology but in a planar form. The conductor that replaces the coaxial line or waveguide is the

stripline [13, 16]. Although literature on the subject varies, this text will refer to the general form of planar transmission line as stripline of which microstrip, slotline and coplanar are sub-categories.

When it comes to planar transmission lines, the microstrip line is most common. Although coplanar waveguide is widely viewed as better than microstrip for most applications, it too has problems. To solve these problems, NASA Lewis and the University of Michigan developed a new version of the coplanar waveguide with electrically narrow ground planes. They called this new transmission line the *finite ground coplanar waveguide* (FGCW).

Through extensive numerical modelling and experimental measurements, they have characterised the propagation constant of the FGCW, the lumped and distributed circuit elements integrated in the FGCW and the coupling between parallel transmission lines. Even though attenuation per unit length is higher for FGCW because of higher conductor loss, it is comparable to conventional coplanar waveguide when the ground plane width is twice that of the centre conductor ($B = 2S$) [18]. This structure is shown in Fig. 4.4.

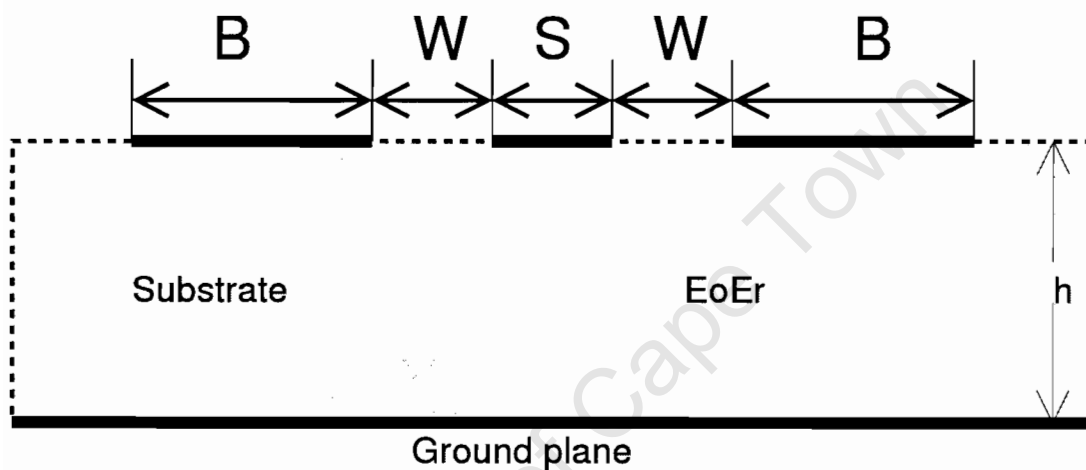


Figure 4.4: Finite ground coplanar waveguide

The first task in designing any form of stripline is to select a board material to suit the intended frequency of operation. For each medium or guiding structure there is an associated permittivity. This quantity indicates how easily a material may be polarised and is expressed as $\epsilon = \epsilon_0\epsilon_r$. Here ϵ_0 is the absolute constant for permittivity and ϵ_r is the relative permittivity, also known as the dielectric constant, of the material used in the MIC.

The second task is to calculate the line width for the intended characteristic impedance. The impedance of a stripline is a function of the dielectric constant, the substrate height h and the line width. As can be seen from Fig. 4.4, the analysis of line width is more complex for FGCW than for that of microstrip. This is because two strips of width B do not exist for microstrip. The reader is referred to [13] for a detailed analysis of impedance calculation. Alternatively, many software options are available for simple calculation of

line widths. These range from simple java applets [21] on one end of the spectrum to complex programs, such as Polar Instruments Si6000 [22], on the higher and more costly side.

Using material with $h = 1.524$ mm, $\epsilon_r = 4.5$ and line dimensions $S = W = 0.5$ mm, $B = 1$ mm produces the desired 50Ω line [21].

With the stripline taken care of, special care should be taken during board layout to minimise parasitics. This is particularly important at frequencies above 1 GHz. The following steps should be adhered to during this process:

- Extra lead length corresponds to extra inductance added to the design. Therefore, effort should be made to ensure components sit flush with transmission lines.
- Abrupt changes in transmission line width also cause parasitic effects called step discontinuities. These can be reduced by tapering transmission lines down to the component lead width.
- Parasitic effects can also be created by bends in a transmission line and hence this should be avoided where possible. If not possible however, the corners should be chamfered to prevent them from acting as extra shunt capacitance.
- Ground planes should be kept solid and as large as possible.
- High frequency return paths must be kept as short as possible. If plated through holes are used as ground returns, they must be placed as near as possible to the body of the package.
- Additional path length acts as series inductance and therefore transmission lines must not be longer than they need to be.

With the FGCW line dimensions set and board guidelines laid out we can now move on to implement the amplifier and frequency synthesizer boards in order to realise the FDU.

4.4 Amplifiers

The amplifiers of the FDU up to 4 GHz are built using monolithic surface mount amplifiers. The circuit of Fig. 4.5 illustrates the biasing configuration and is obtained from the data sheet of the particular amplifier. FGCW will connect the 50Ω components between the RF input and output. These include two DC blocking capacitors and the amplifier. The biasing resistor is calculated with a supply of +9 volts, and knowledge of the biasing current. A RF choke of large impedance is inserted to combat the "power divider" effect, where the bias resistor appears in parallel with the output load impedance.

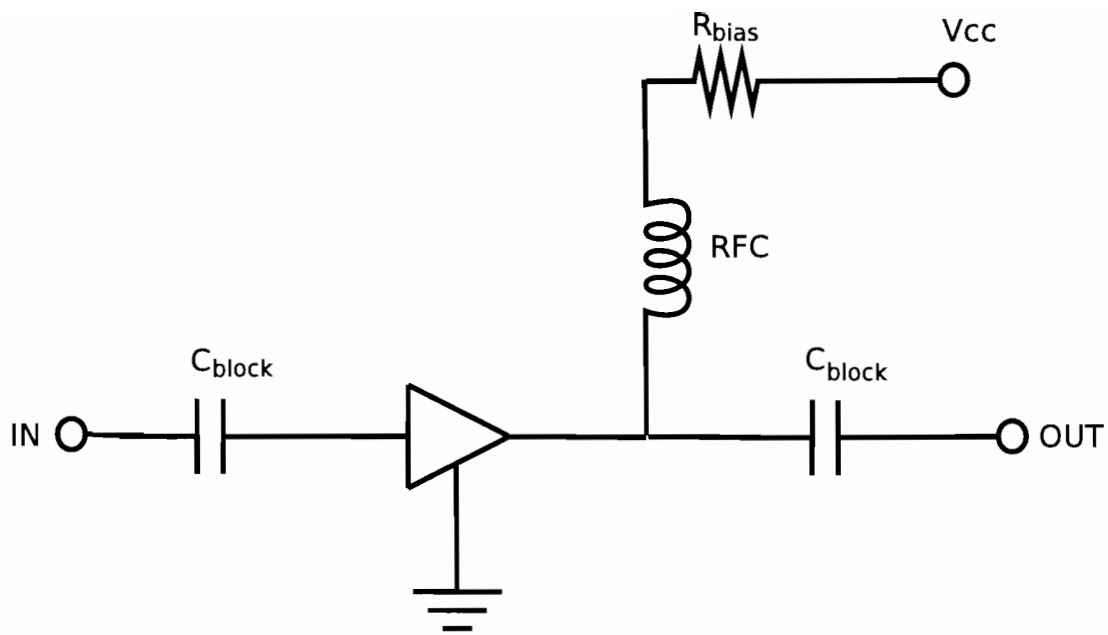


Figure 4.5: Typical biasing configuration for Mini-circuits ERA amplifiers.

The boards are laid out using Protel and the the PCBs are acquired. The components are then assembled and appropriate SMA connections are added. The boards are now realised as shown in Figs. 4.6 and 4.7. The amplifiers connecting the the six way splitter to the synthesizers are included on the synthesizer board to reduce the component count. All boards are equipped with regulators to keep check on the supply voltage.

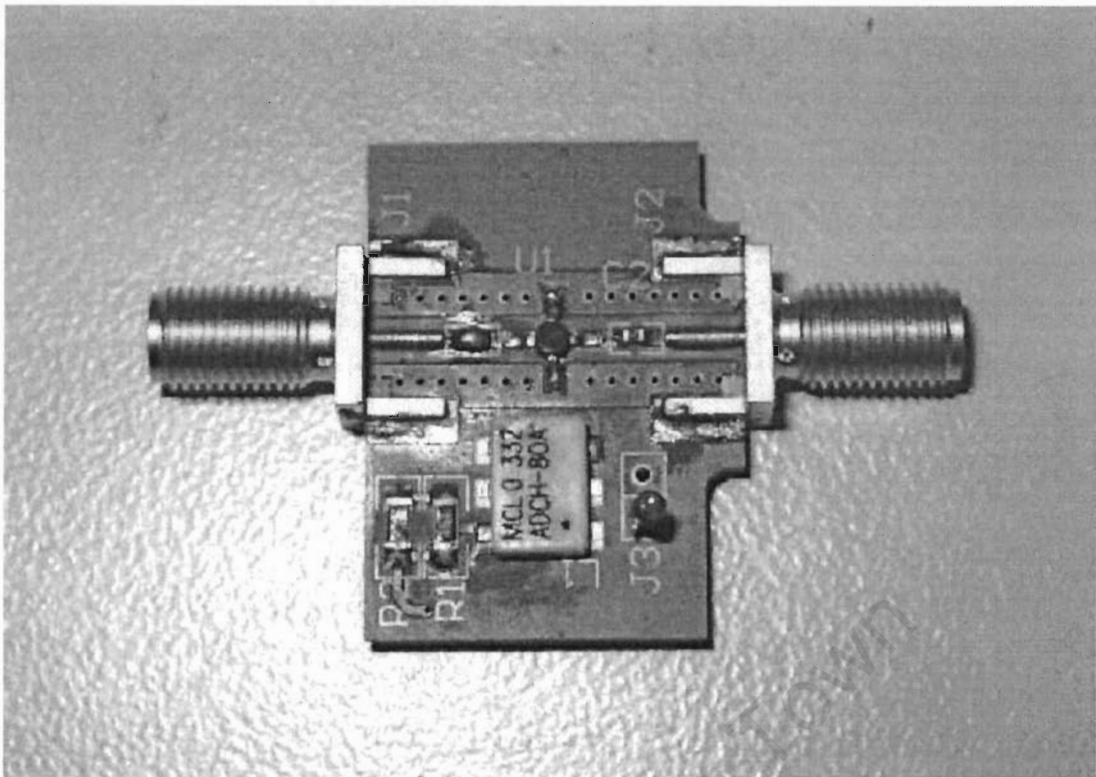


Figure 4.6: Realisation of board layout for the Mini-circuit's ERA amplifiers.

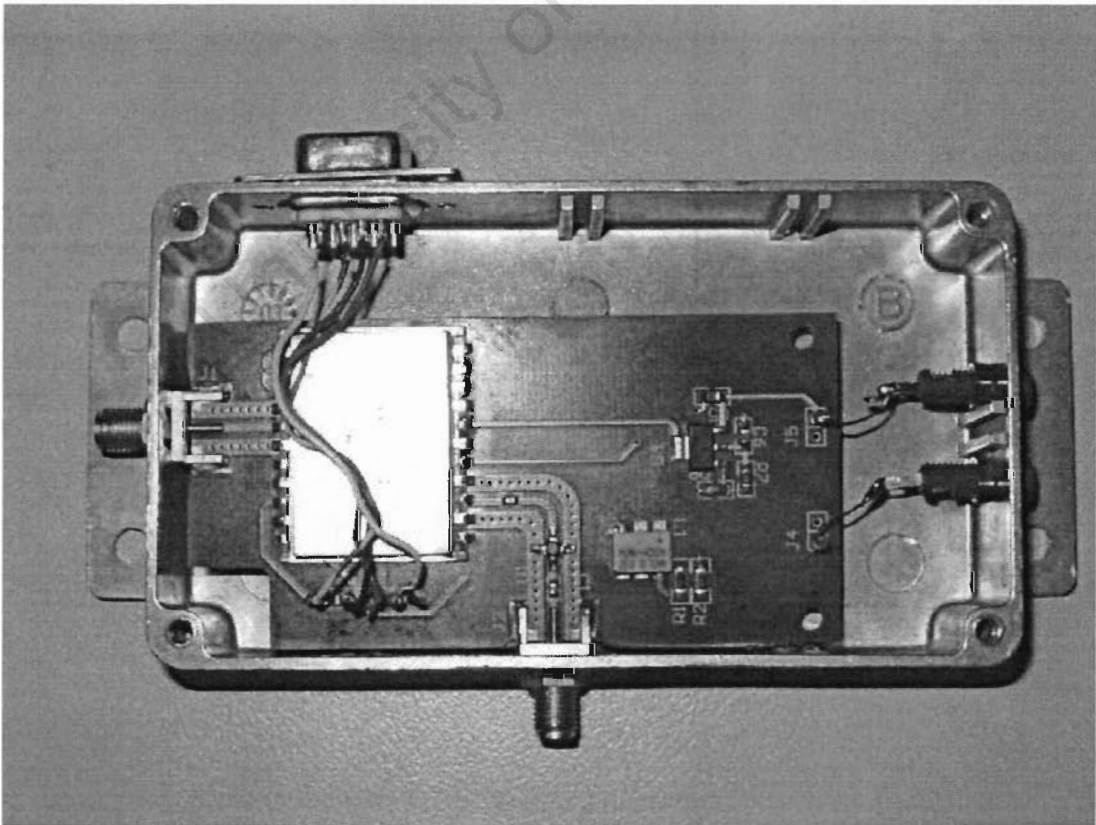


Figure 4.7: Realisation of board layout for the Synergy frequency synthesizers.

4.5 Microwave Housing Design

Microwave circuits are usually placed in a metal housing with connectors through the walls onto the appropriate stripline of the circuit. The main purpose of the housing is to provide mechanical strength and handling, electrical shielding and potentially heatsinking. A problem may arise in that a housing with a lid will form a cavity, and thus exhibit resonance.

With the stripline circuit in the housing, this may be looked upon as a dielectrically loaded cavity resonator. There is a possibility of performance deterioration if the lowest order cavity mode were to couple to the stripline circuit. The problem, frequently, is that the lowest order cavity mode sets a limitation on the maximum frequency for satisfactory operation of the stripline circuit [16].

We shall now attempt to calculate the cavity mode for the given housing dimensions. Literature [23] states that the added complexity of using more complicated equations to calculate the cut-off frequency of the cavity with substrate inside are not worth the extra effort. This allows us to assume a dielectric constant of 1 (for air). Thus:

$$f_{10} = \frac{\frac{15}{w} \sqrt{1 + (\frac{w}{l})^2}}{\sqrt{\epsilon_r}}$$

where l and w are in cm and f is in GHz.

With the available housing for amplifiers, where $w = 3.5$ cm and $l = 8.9$ cm, the cavity mode is set to $f_{10} = 4.6$ cm.

For the synthesizer housings, where $w = 6.4$ cm and $l = 11.4$ cm, the cavity mode is set to $f_{10} = 2.7$ GHz.

It is advisable to use a housing with cavity mode at least 3 times that of the stripline circuit. However, if problems exist, microwave absorber material may be added to the cavity to provide a quick and inexpensive solution to cavity resonances [23, 24].

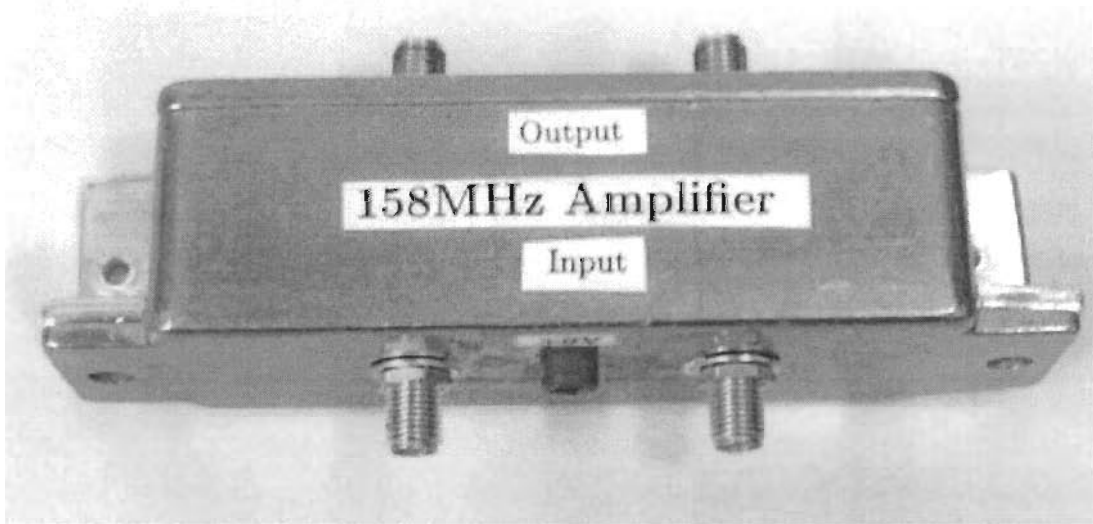


Figure 4.8: Completed module for the 158 MHz amplifiers (AMP11-2 and AMP11-3).



Figure 4.9: Completed module for the 150 MHz frequency synthesizer (SZ4).

4.6 Phase Noise and Jitter

Now that the FDU has been designed and implemented, a measure of the fidelity of the produced signals must be established. This measure is termed phase noise in the fre-

quency domain or jitter in the time domain.

Phase noise is defined as the undesired phase variation of a signal. This is indicated by the spreading of the frequency spectrum ($s(t) = A\sin(\omega t + f(t))$), as opposed to an ideal signal ($s(t) = A\sin(\omega t)$) where the power is contained at one specific frequency. A common measure of short-term frequency stability is the parameter $L(f_m)$, which refers to the *one-sided power spectral density of phase noise*. As can be seen in Fig. 4.10, $L(f_m) = N/C$ is the difference of power between the carrier at f_0 and the noise at $f_0 + f_m$. The spectral distributions on either side of the carrier are known as noise sidebands. The units of $L(f_m)$ are decibels below the carrier (dBc) per hertz .

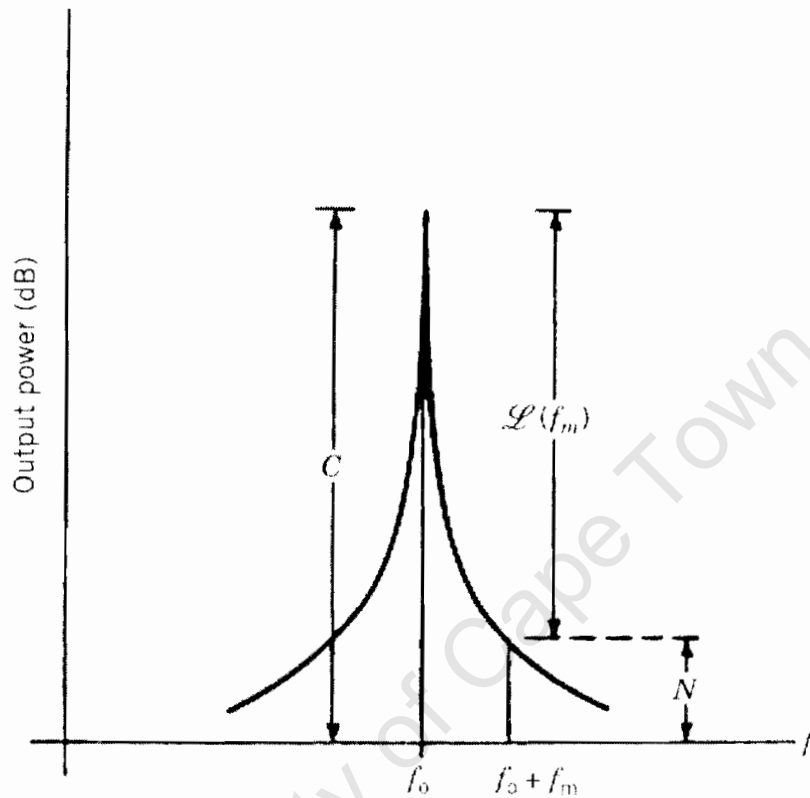


Figure 4.10: Oscillator output power spectrum [15].

The bulk of oscillator noise close to the carrier is phase noise. This noise represents the phase jitter or short-term stability of the oscillator. Jitter and phase noise are different ways of quantifying the same phenomenon. The difference being that one is a time domain measurement, while the other is a frequency domain measurement [15, 19, 20].

4.7 Summary

Frequency synthesizers were discussed. It determined that they would provide an efficient means of generating coherent signals for the clocks and LO drives. Synergy components

were selected. The FDU was attained by working power levels around the synthesizers with knowledge of the reference output power. This detailed the required amplifiers. A breakdown of the components of the FDU appears below (see Fig. 4.3 for component names):

Table 4.1: Components used in the FDU.

Component name	Manufacturer	Part number
AMP11-1	Mini-circuits	ERA-50SM
AMP11-2	Mini-circuits	ERA-4SM
AMP11-3	Mini-circuits	ERA-4SM
AMP12-1	Mini-circuits	ERA-50SM
AMP12-2	Mini-circuits	ERA-2SM
AMP12-3	Mini-circuits	ERA-4SM
AMP13-1	Mini-circuits	ERA-50SM
AMP13-2	Mini-circuits	ERA-4SM
AMP13-3	Tellumat	Custom
AMP13-4	Tellumat	Custom
AMP14-1	Mini-circuits	ERA-50SM
AMP15-1	Mini-circuits	ERA-50SM
FD	Mica	F15KSP
FL9	Lorch	Custom
SP3	Mini-circuits	ZSBC-615
SP4	Mini-circuits	ZMSCQ-2-180
SP5	Mini-circuits	ZFSC-2-5
SP6	Mini-circuits	ZFSC-2-9-G
SZ1	Synergy	SPLL-S-A40
SZ2	Synergy	SPLH-S-A79
SZ3	Synergy	SPLH-4000F
SZ4	Synergy	SPLL-S-A40
SZ5	Synergy	SPLL-S-A40

Theory relevant to the design of RF PCBs was discussed and, with this knowledge, the synthesizer and amplifier boards were constructed. They were housed in metal casings which were analysed to determine if resonance will hinder the circuits performance. Phase noise and jitter were introduced with the aim of understanding how to quantify the effectiveness of the signals produced.

Chapter 5

Testing

5.1 Introduction

Testing of the transmitter and FDU is performed in order to verify the design. The following equipment is used:

Table 5.1: Bench equipment used for testing.

Equipment	Model
DC power supply	Topward 6303A
Digital multimeter	Escort EDM-2116L
Spectrum analyzer	Agilent E4407B (9kHz-26.5GHz)
Sweep oscillator	Hewlett Packard 8350B
Power meter	Hewlett Packard HP 435A
Arbitrary waveform generator (AWG)	Hewlett Packard 33120A

Where appropriate, similar FDU and transmitter components are tested together. However, the tests follow the general trend of FDU first, followed by transmitter. This results from the necessity of using the FDU for various transmitter tests.

Thus, amplifiers are tested first (from both FDU and transmitter). The synthesizers are then tested and a power budget for the FDU is drawn up. The first IF is then tested with the use of the FDU to inspect the integrity of the chirp signal produced. A CW signal is injected to the transmitter input with the aim of finalising the transmitter power budget. A plot of the final chirp output spectrum is then compared to one of the the input to establish the distortion of the signal's magnitude through the system. The findings are then discussed after all tests have been done and results shown.

5.2 Amplifier Testing

5.2.1 Equipment Used

DC power supply, power meter, sweep oscillator.

5.2.2 Test Procedure

The sweep oscillator is connected via a coaxial test cable to the power meter. The sweep oscillator is set to continuous wave (CW) mode and adjusted until the power meter reads the desired input power level. The test cable is then disconnected and the FDU amplifier boards are individually inserted with an attenuator equal to their gain. This process serves to exclude losses outside the test circuit boundaries from the measurements and leave the power meter settings unchanged. The DC supply is then attached to the amplifier and the output is read off the power meter.

The procedure for the two transmitter amplifiers (Amps 1&2) is slightly different in that the signal passed through them is of 100 MHz bandwidth. This requires the sweep oscillator to be changed from CW to sweep mode. This is applied to the spectrum analyzer and the hold maximum button is pushed to record the trace of the signal. A screen snapshot is taken. Thereafter, the amplifier and an attenuator equal to the value of the amplifier gain is inserted. This allows the spectrum analyzer settings to remain unchanged. The DC supply is attached and the hold maximum button is pushed. A snapshot is again recorded.

5.2.3 Test Results

Table 5.2: Test results for the various amplifier boards.

Amplifier	Frequency (MHz)	Input (dBm)	Output (dBm)	Gain (dB)	Expected Gain (dB)
Amp11-2	158	+2.0	+12.5	10.5	11.0
Amp11-3	158	+2.0	+12.4	10.4	11.0
Amp12-2	1142	0.0	+13.4	13.4	13.0
Amp12-3	1142	0.0	+11.2	11.2	11.0
Amp13-2	4000	+2.0	+13.0	11.0	11.0
Amp13-3	8000	-2.0	+13.8	15.8	16.0
Amp13-4	8000	-2.0	+13.8	15.8	16.0

Refer to Fig. 4.3 for the amplifier's name designation.

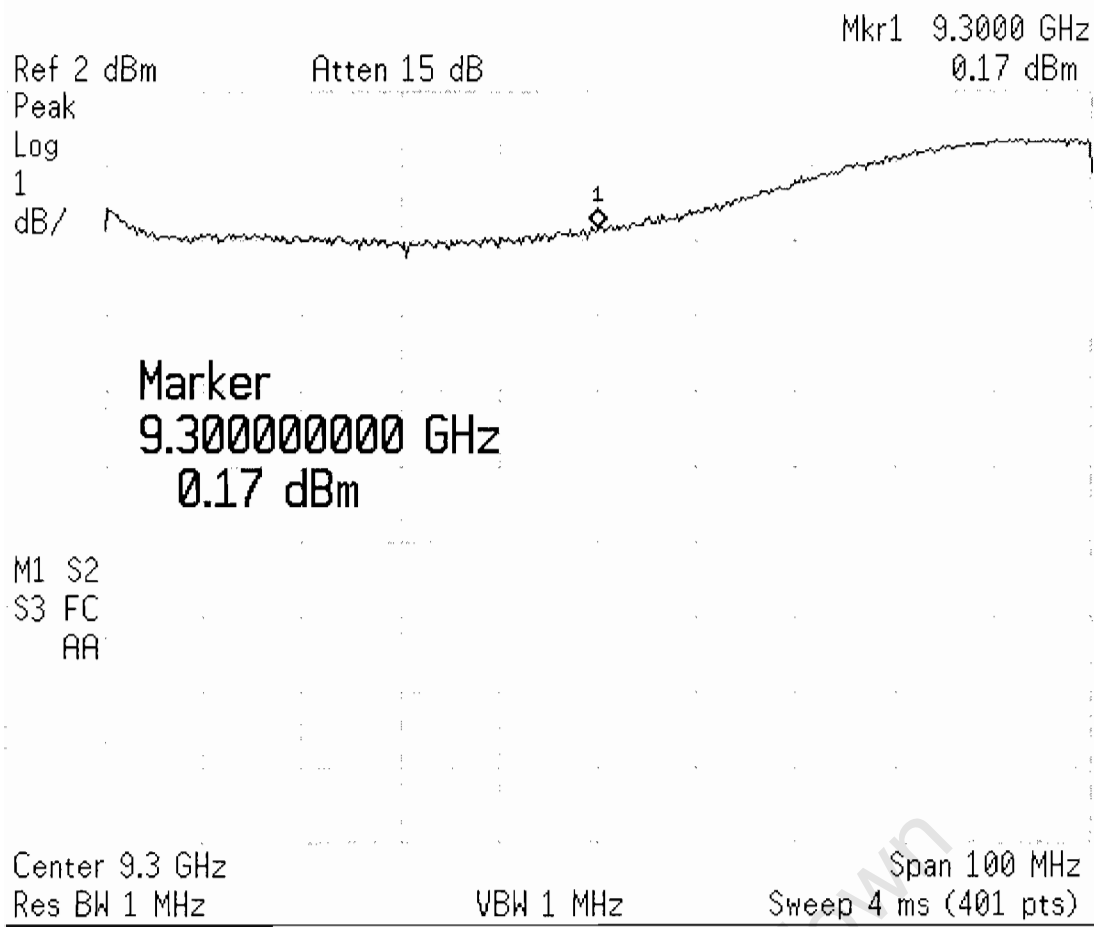


Figure 5.1: Input to transmitter amplifier.

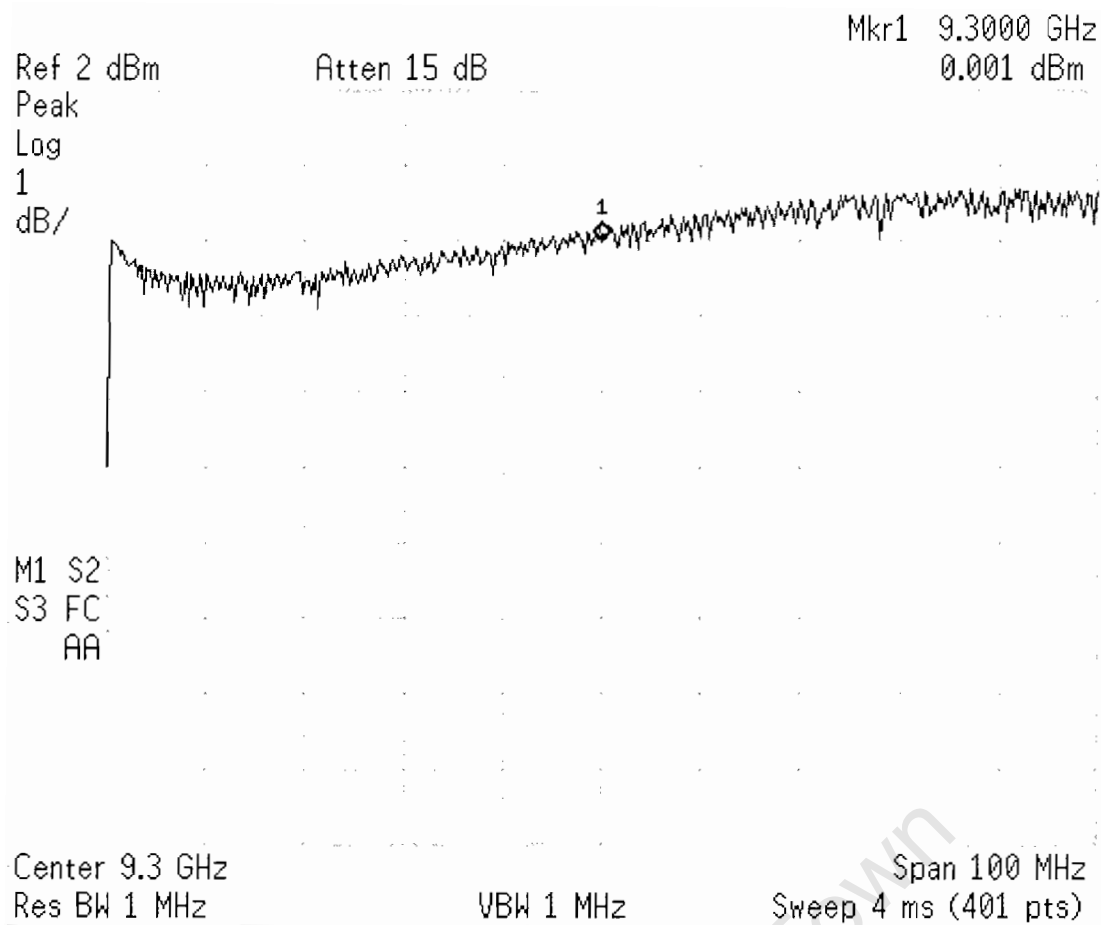


Figure 5.2: Output of transmitter amplifier with a 20 dB attenuation.

By subtracting Fig. 5.2 from Fig. 5.1, one can get an idea of how the amplifier maintains its response over the frequency band of interest. This is shown in Table 5.3 below.

Table 5.3: Amplifier response over the 100 MHz bandwidth.

Frequency (GHz)	8.8	8.9	9.0	9.1	9.2	9.3	9.4	9.5	9.6	9.7	9.8
Amplitude (dB)	0.4	0.6	0.6	0.3	0.1	0.1	0.2	0.4	0.6	0.7	0.7

5.3 Synthesizer Testing

5.3.1 Equipment Used

DC power supply, spectrum analyzer, arbitrary waveform generator (AWG).

5.3.2 Test Procedure

The frequency and power level of each synthesizer is ascertained by the following process: the AWG is set to 10 MHz at +4 dBm. The synthesizer output is connected to the spectrum analyzer and the power is supplied. The serial bit stream is then transferred from Codeloader2 for the synthesizer of concern. The frequency and power level is recorded.

In order to obtain a phase noise plot, the spectrum analyzer must be changed to phase noise mode. The carrier frequency and start (10 Hz) and stop (1 MHz) points must be input. This results in a plot being generated.

5.3.3 Test Results

Table 5.4: Synthesizer's output frequency and power level.

Synthesizer name	Frequency (MHz)	Output (dBm)	Expected Output (dBm)
SZ1	157.999	+5.42	+5.0
SZ2	1141.999	+4.70	+5.0
SZ3	3999.998	+3.08	+3.0
SZ4	149.999	+5.27	+5.0
SZ5	209.999	+5.43	+5.0

The graphs of each synthesizer output spectrum are displayed in Appendix B. These phase noise plots follow:

Carrier Freq 3.999997736 GHz

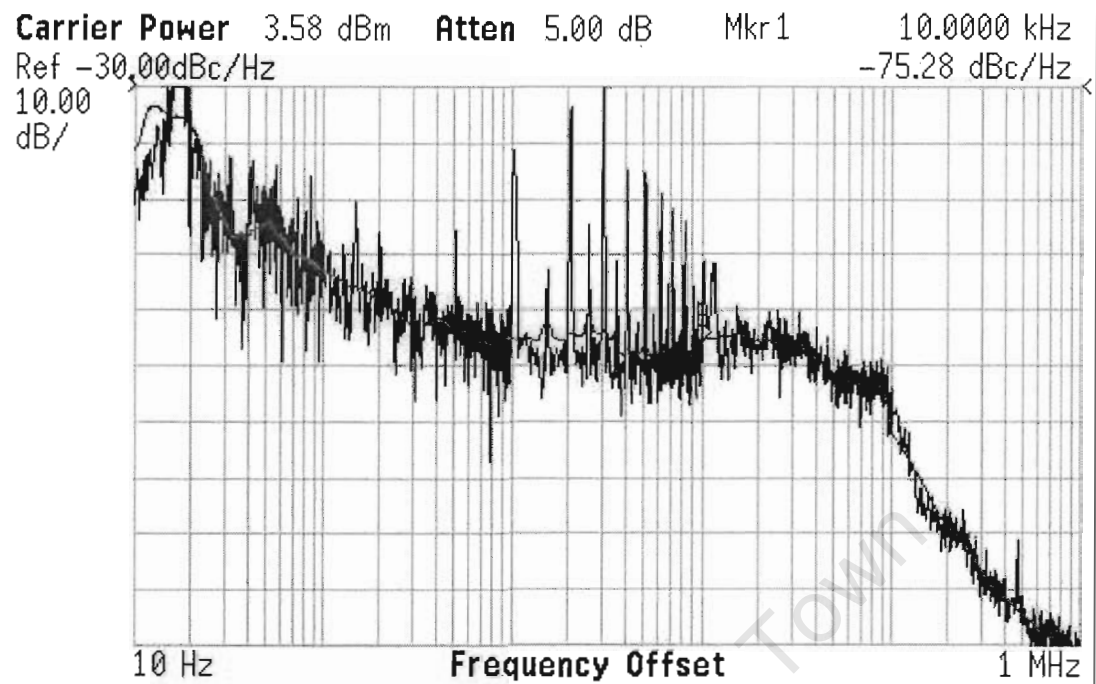


Figure 5.5: Phase noise plot for the 4000 MHz synthesizer.

Carrier Freq 149.9999230 MHz

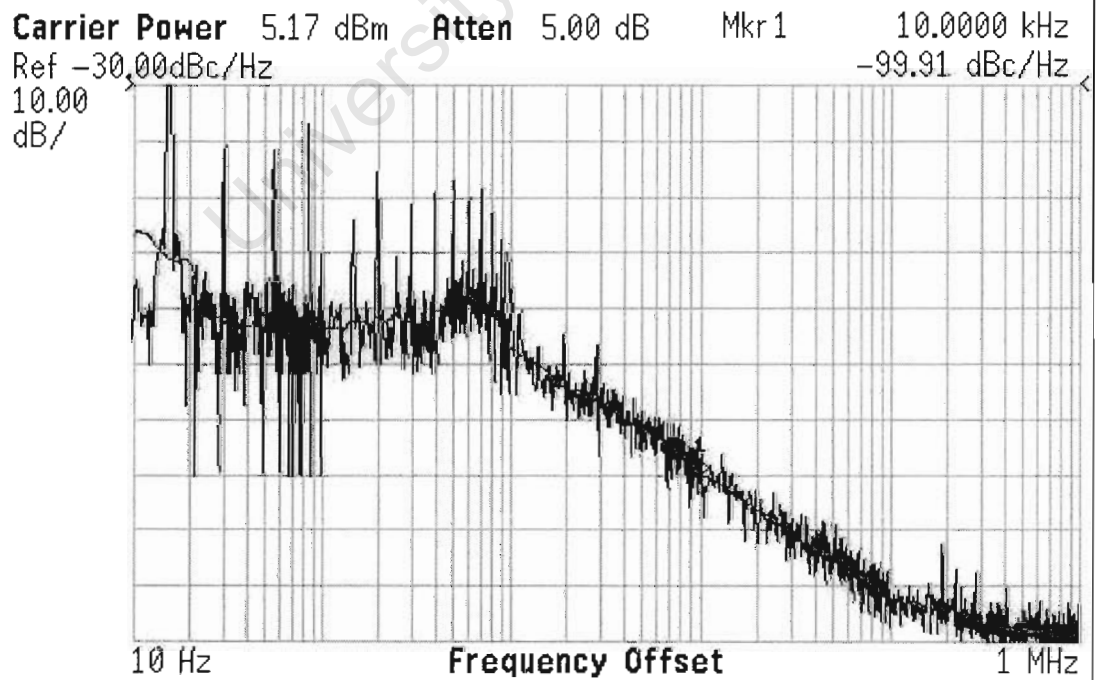


Figure 5.6: Phase noise plot for the 150 MHz synthesizer.

5.4.3 Test Results

Table 5.5: Measured output power of the FDU.

Arm frequency (MHz)		Output Power (dBm)	Calculated Output (dBm)
158	I	+12.5	+12.7
	Q	+12.4	+12.7
1142	T	+14.4	+12.2
	R	+12.2	+10.2
4000	T	+12.1	+13.8
	R	+12.1	+13.8
150	-	+5.27	+5.0
210	-	+5.43	+5.0

5.5 Power Levels of the Transmitter

5.5.1 Equipment Used

DC power supply, power meter, sweep oscillator, AWG.

5.5.2 Test Procedure

The sweep oscillator is connected via a coaxial test cable to the power meter. The sweep oscillator (in CW mode) is then adjusted to the correct power level. The first IF section is then inserted (with appropriate FDU LO) and the DC supply attached and synthesizer programmed. The power is then read off the power meter. This process is repeated with IF2 added, and again with RF (the whole transmitter). The final two RF amplifiers are not included.

5.5.3 Test Results

Table 5.6: Measured transmitter power levels.

Input Power (dBm)	IF1 Power (dBm)	IF2 Power (dBm)	RF Power (dBm)
0dBm	+1.1	-9.2	-17.1

5.6 Chirp Integrity

5.6.1 Equipment Used

DC power supply, spectrum analyzer, sweep oscillator, AWG.

5.6.2 Test Procedure

Each DPG output is connected successively to the spectrum analyzer and recorded. They are then, along with the the two LO signals from the FDU, connected to the IF1 mixers. Power is applied to the 158 MHz synthesizer and amplifiers. The programming data is then sent to the synthesizer. The spectrum analyzer is attached to the IF1 filter output and a trace is recorded. This process is repeated with IF2 added, then again with the addition of RF.

5.6.3 Test Results

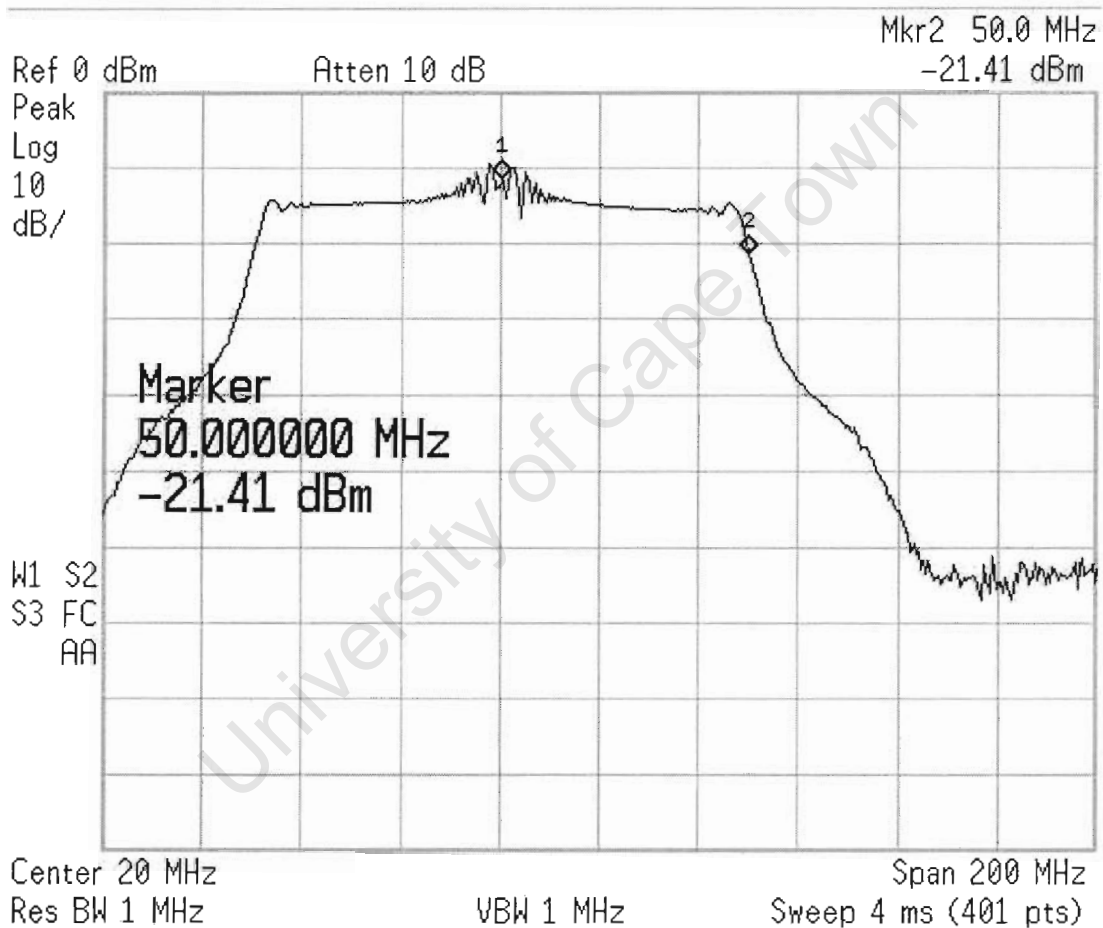


Figure 5.8: Spectrum of baseband in-phase chirp.

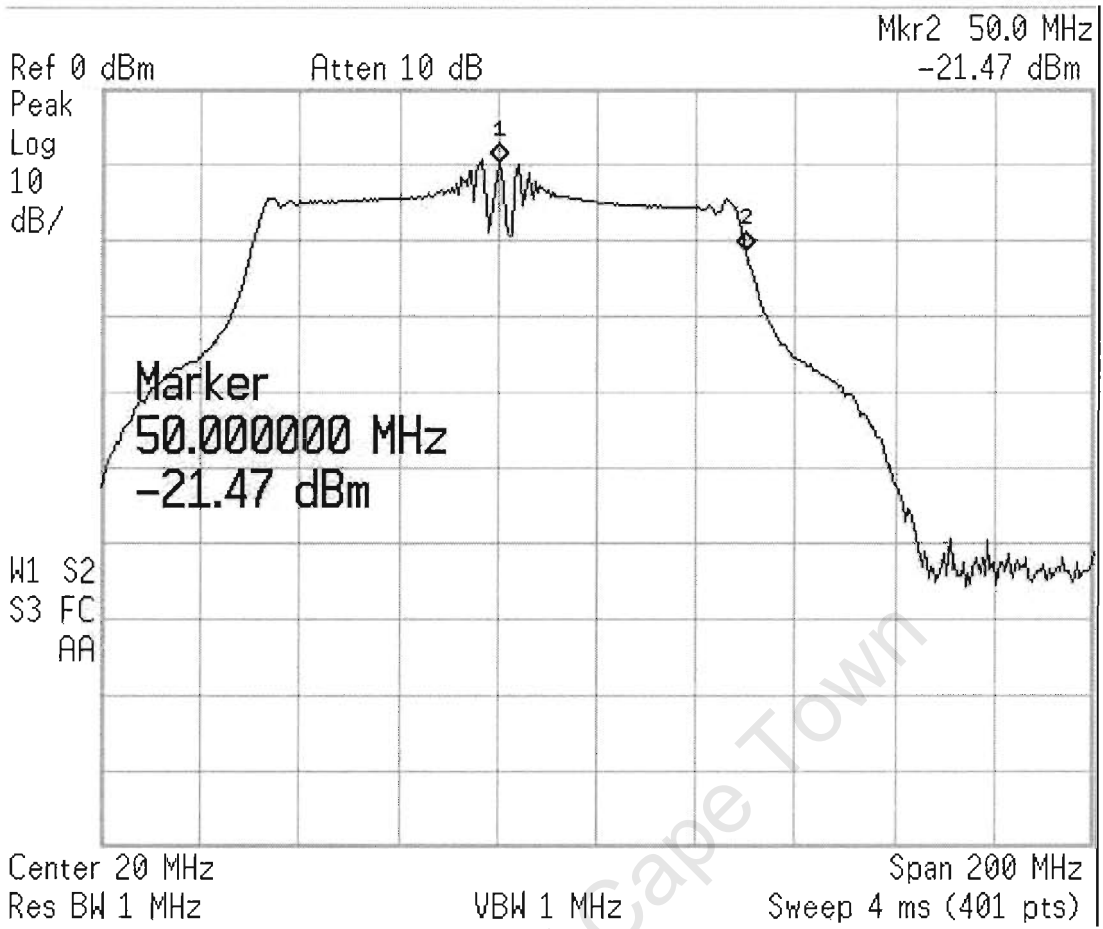


Figure 5.9: Spectrum of baseband quadrature chirp.

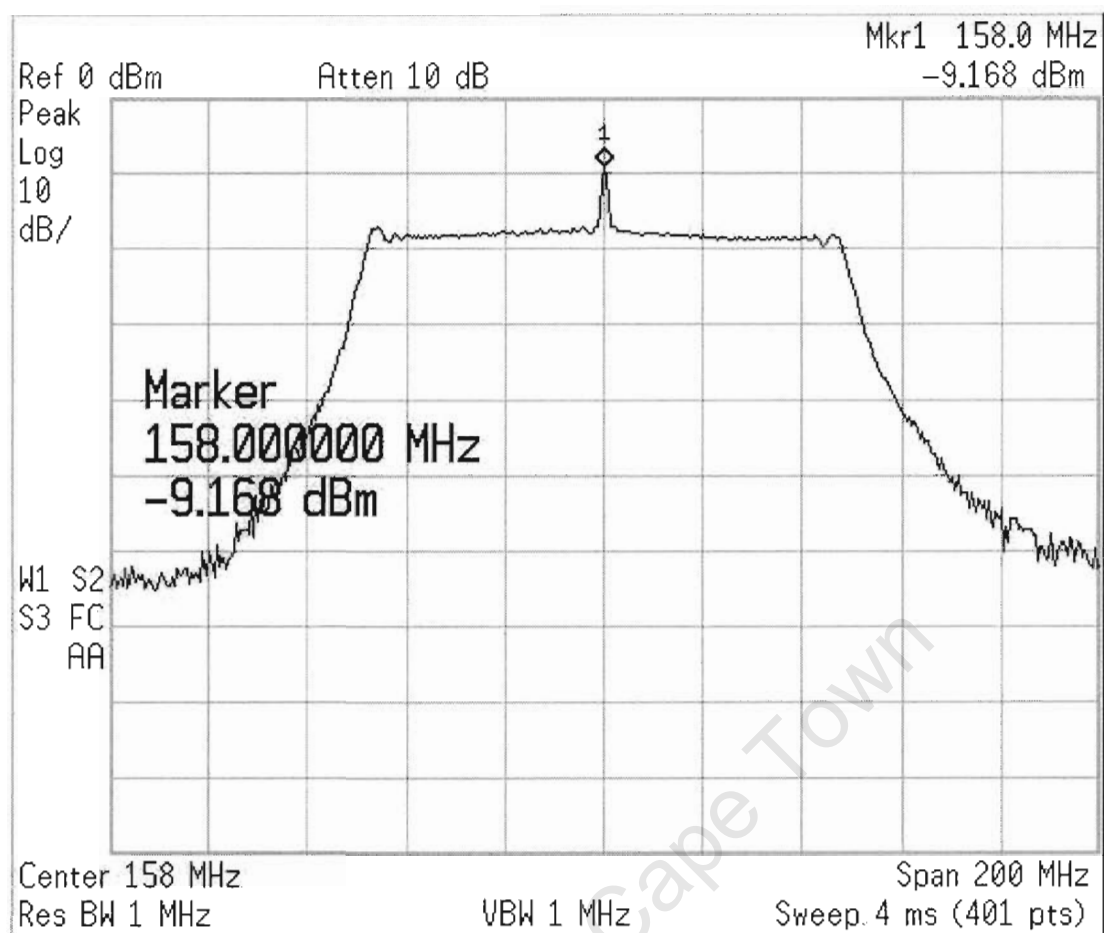


Figure 5.10: Spectrum of 158 MHz chirp.

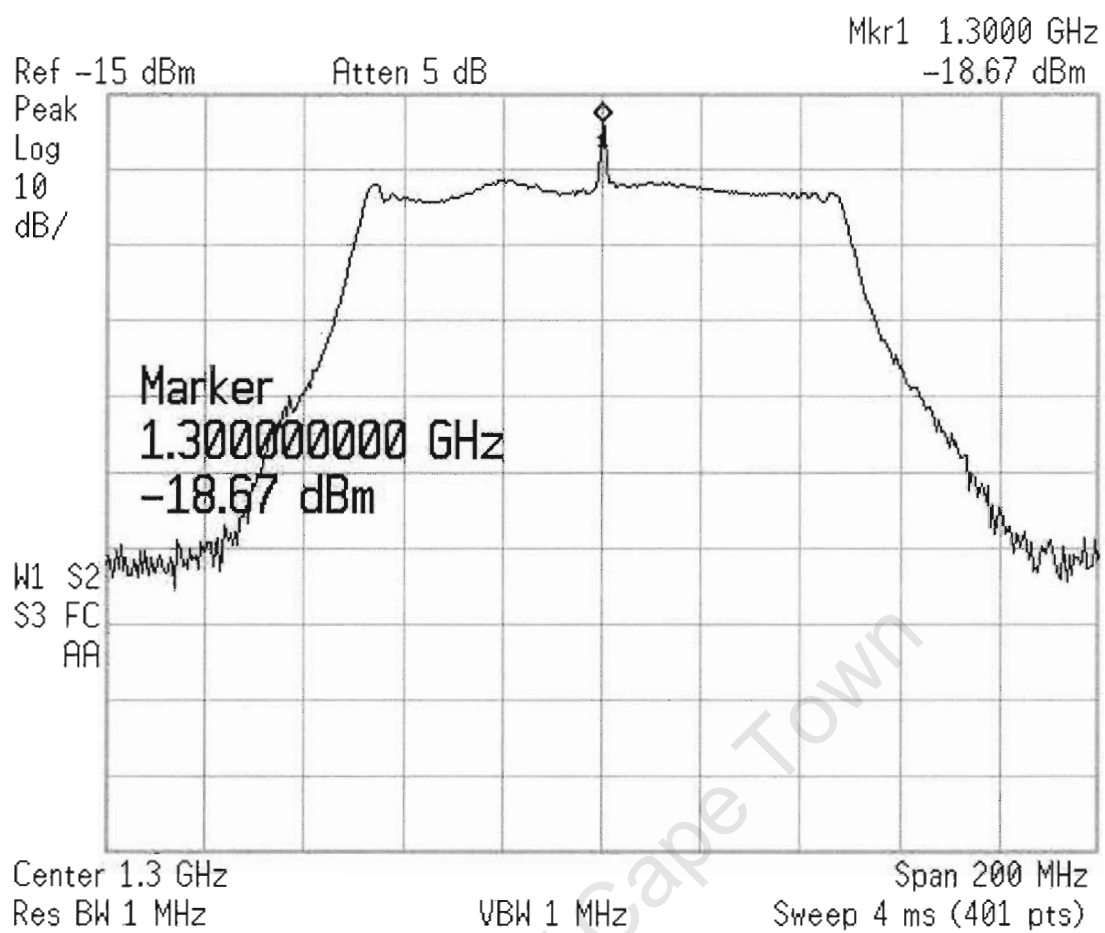


Figure 5.11: Spectrum of 1300 MHz chirp.

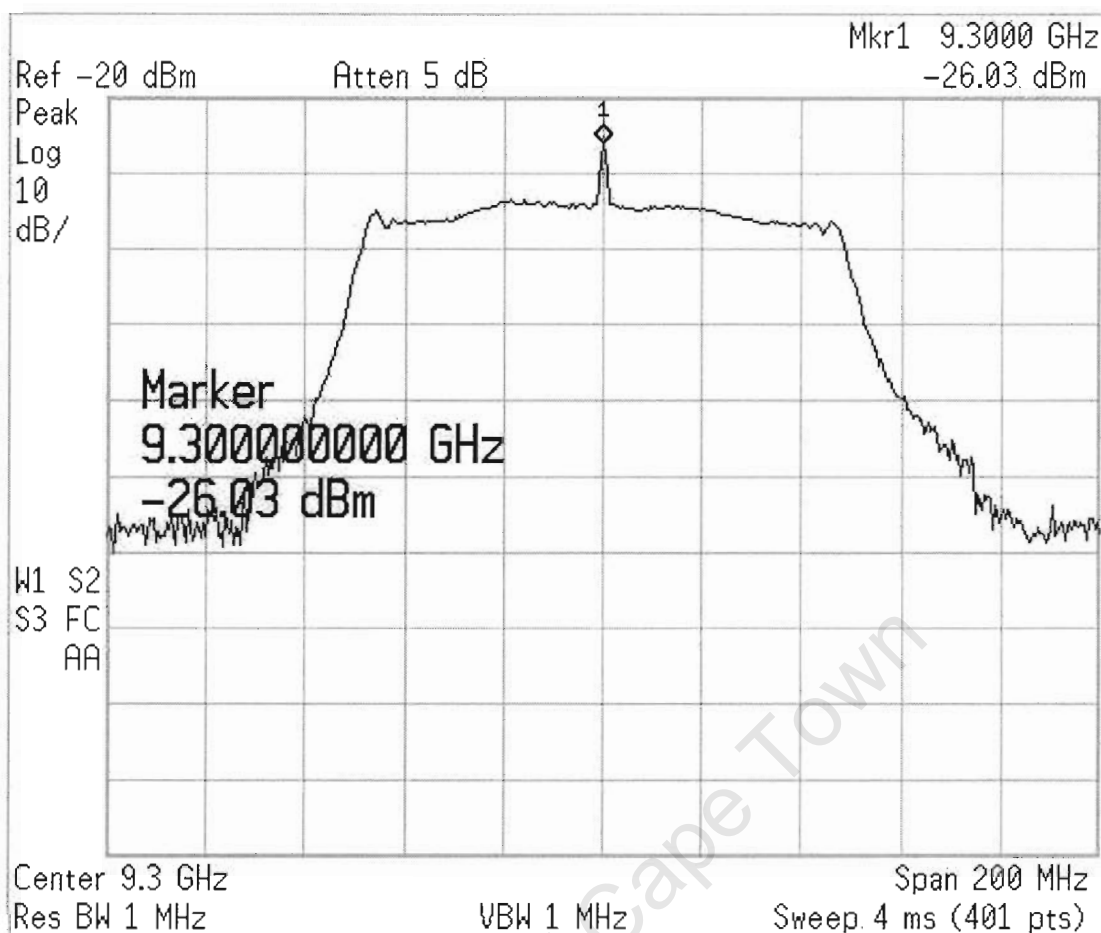


Figure 5.12: Spectrum of 9300 MHz chirp.

5.7 Spurious Signals

5.7.1 Equipment Used

DC power supply, spectrum analyzer, AWG.

5.7.2 Test Procedure

The first IF section is connected (with appropriate FDU LO) and DC supply attached. The synthesizer is programmed and the DPG applied to the input. The video and resolution bandwidths are reduced to 3 kHz on spectrum analyzer. The resultant chirp signal is then recorded. The chirp signal is turned off with the DPG still attached and another trace is plotted. This process is repeated with IF2 added, and again with RF (the whole transmitter and FDU).

5.7.3 Test Results

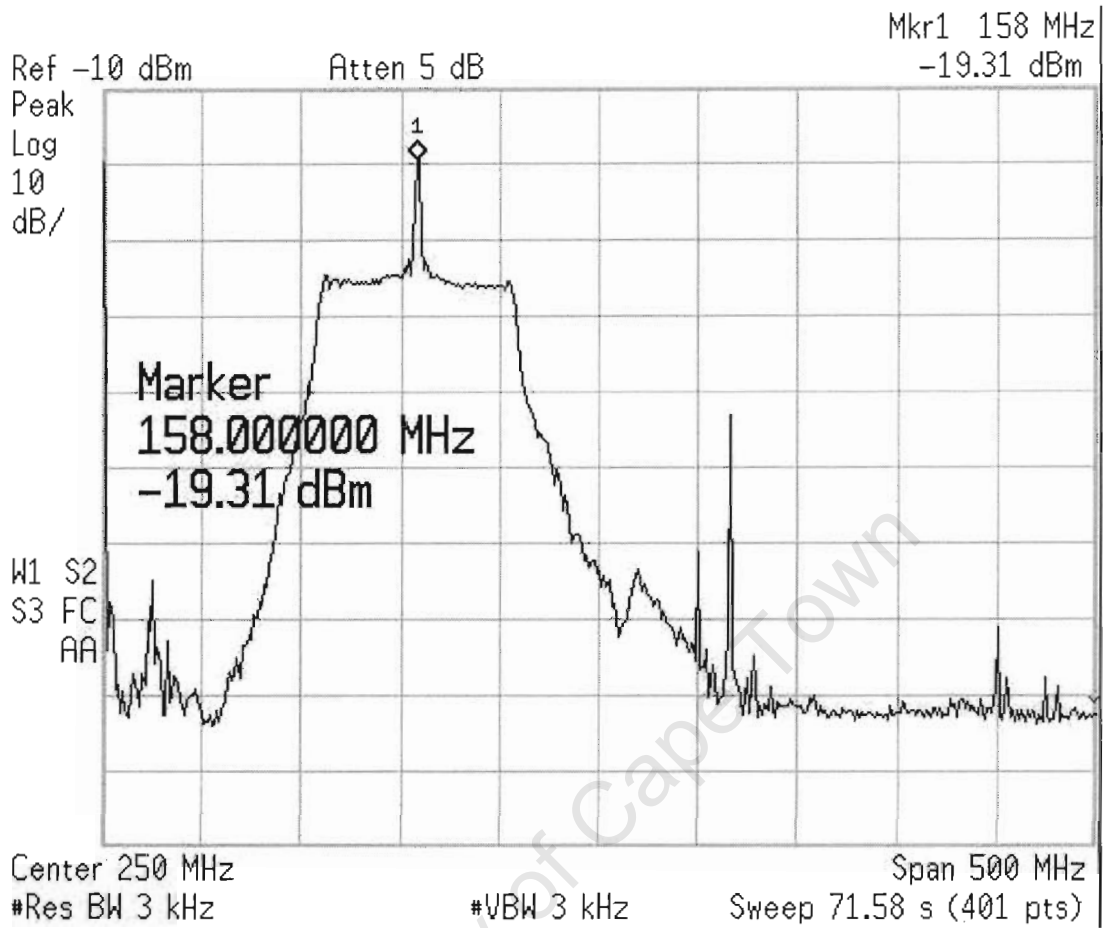


Figure 5.13: Spectrum of 158 MHz chirp with video/resolution bandwidth reduced.

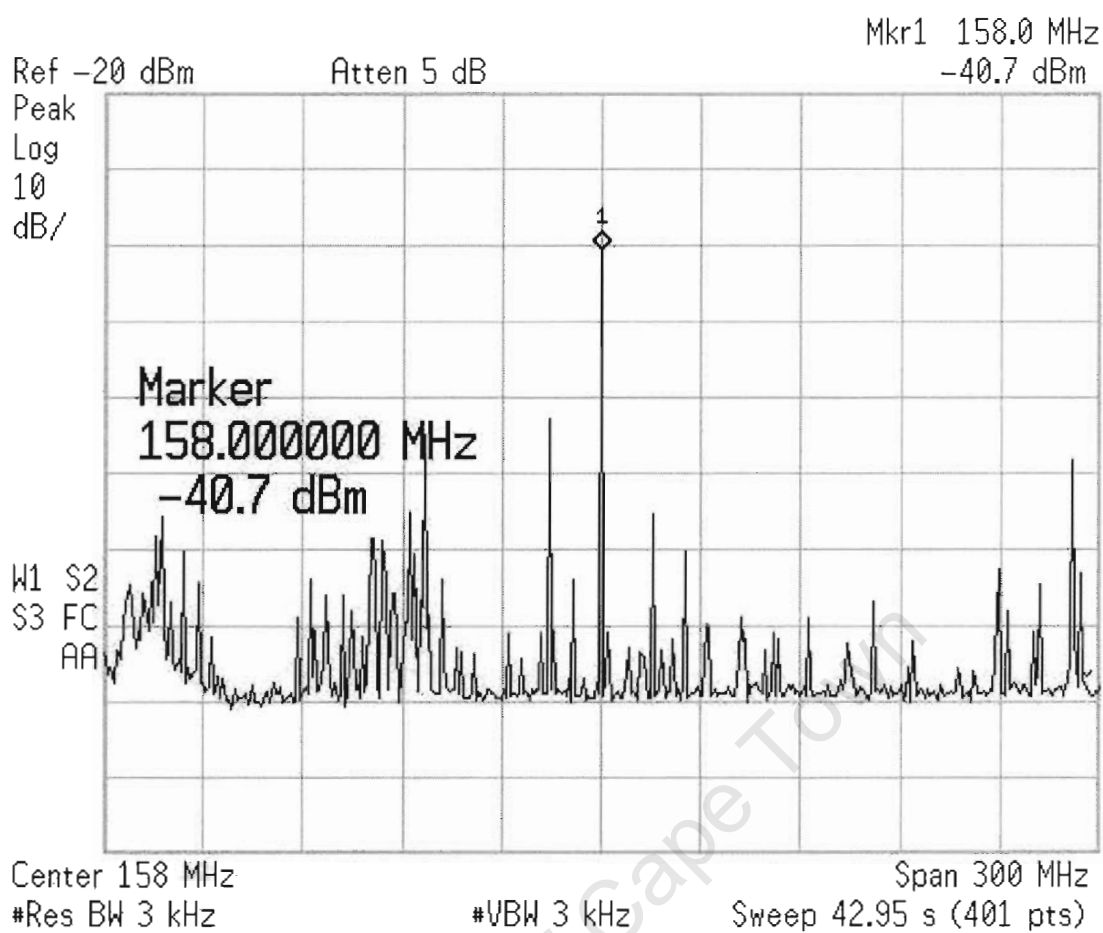


Figure 5.14: Spectrum without the presence of the 158 MHz chirp.

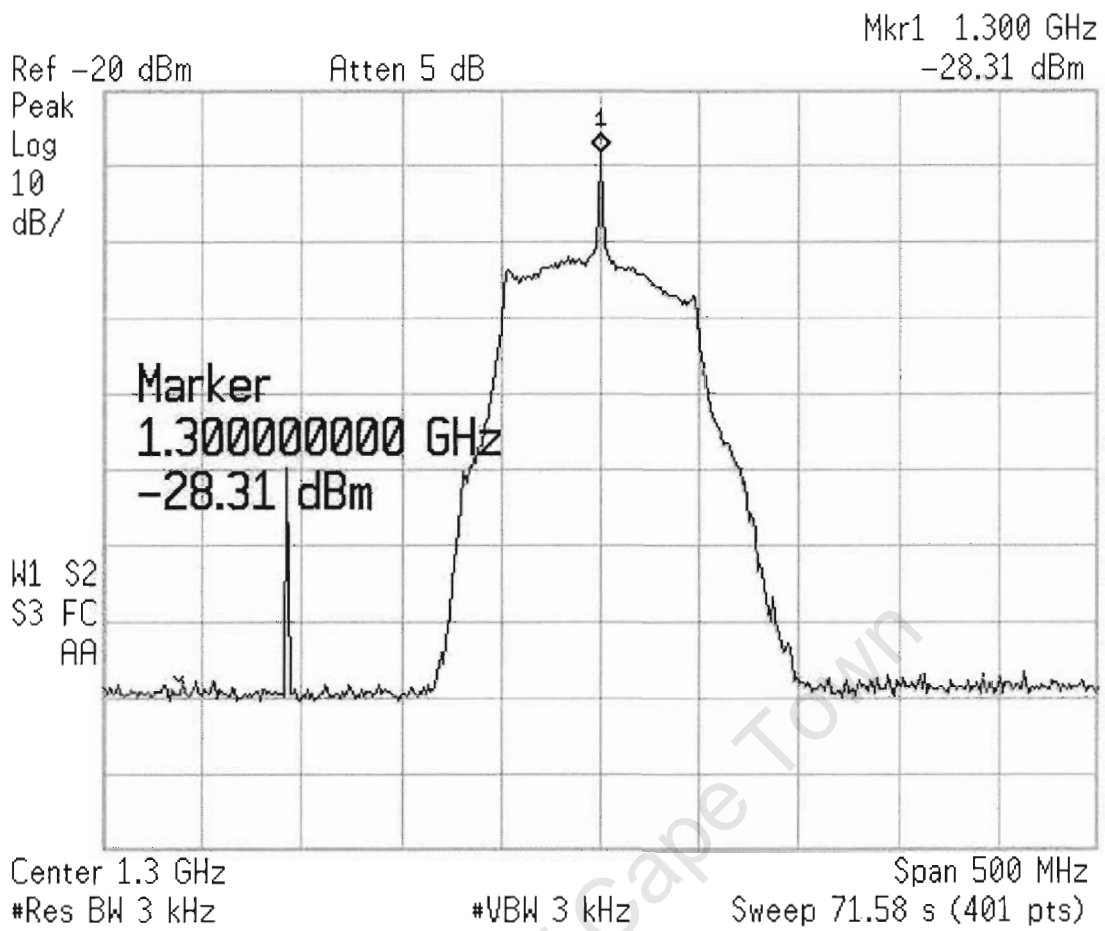


Figure 5.15: Spectrum of 1300 MHz chirp with video/resolution bandwidth reduced.

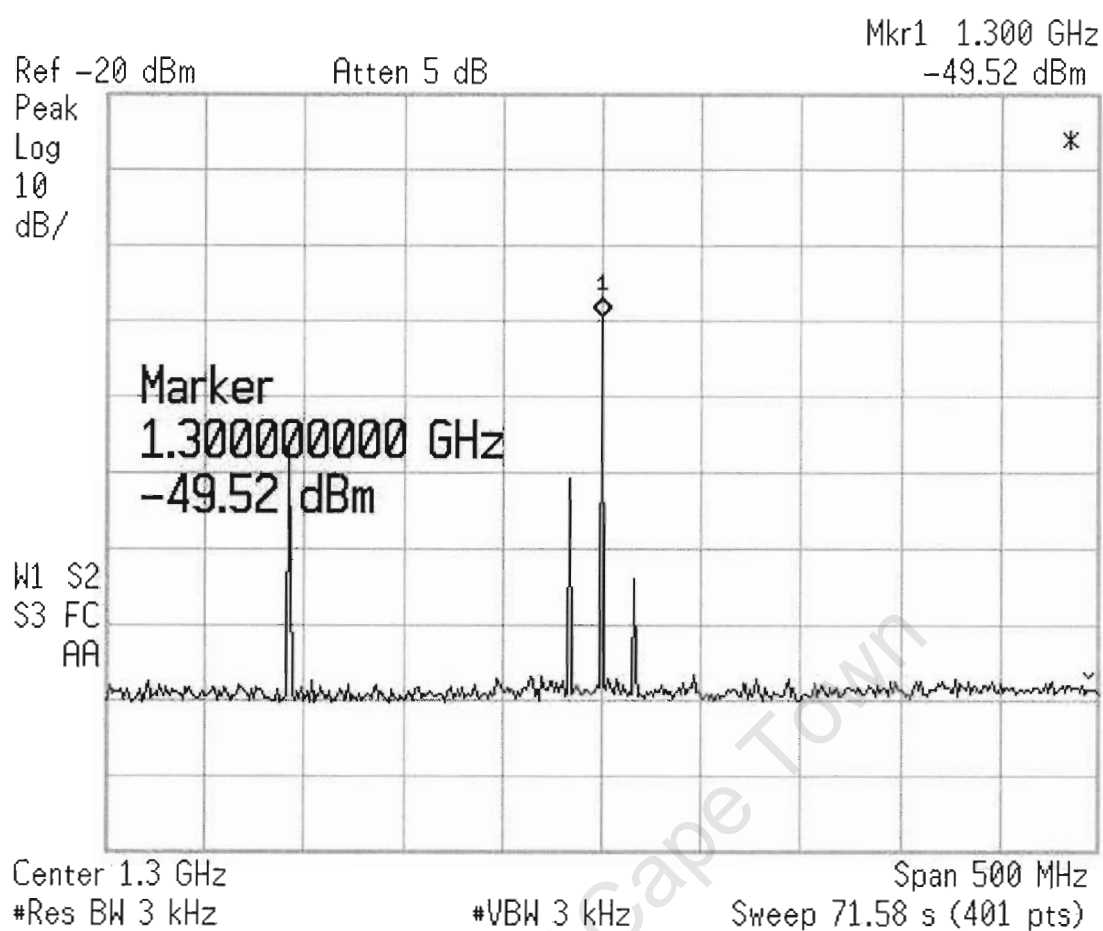


Figure 5.16: Spectrum without the presence of the 1300 MHz chirp.

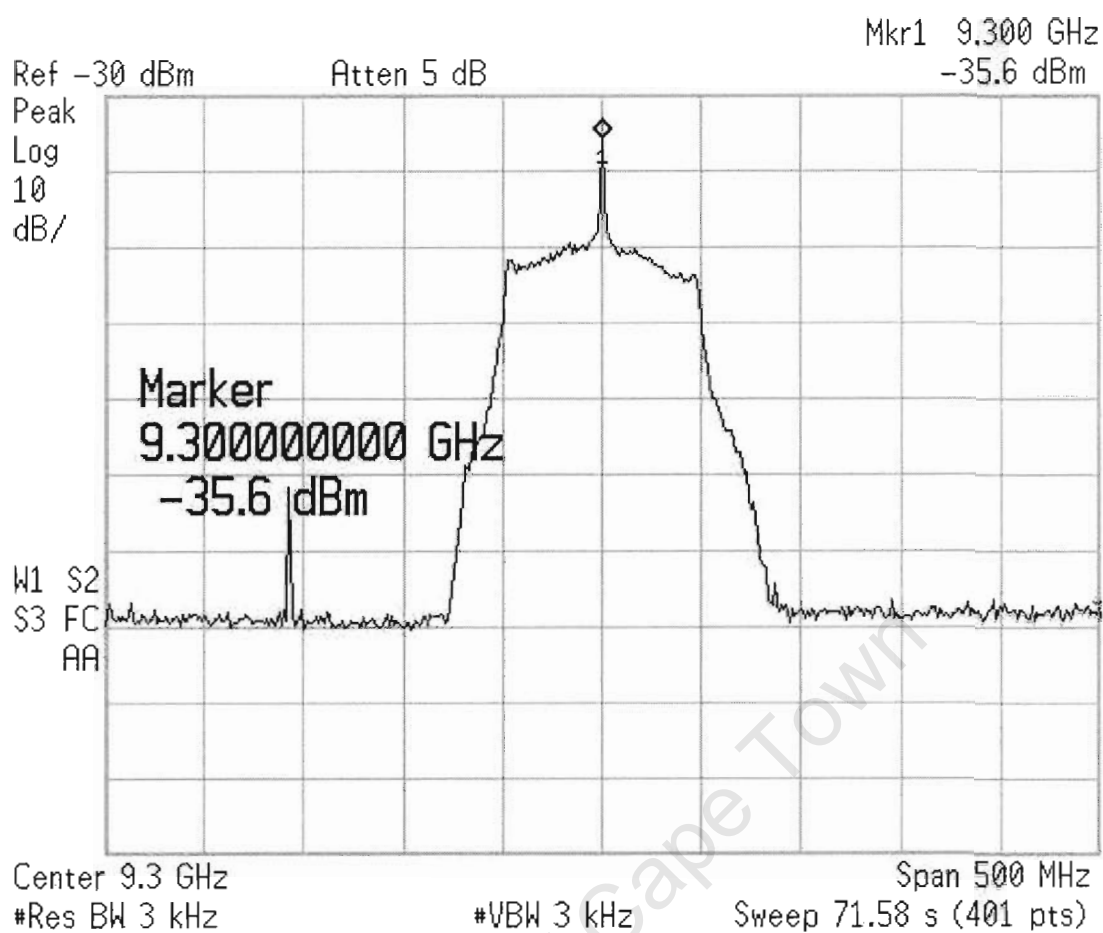


Figure 5.17: Spectrum of 9300 MHz chirp with video/resolution bandwidth reduced.

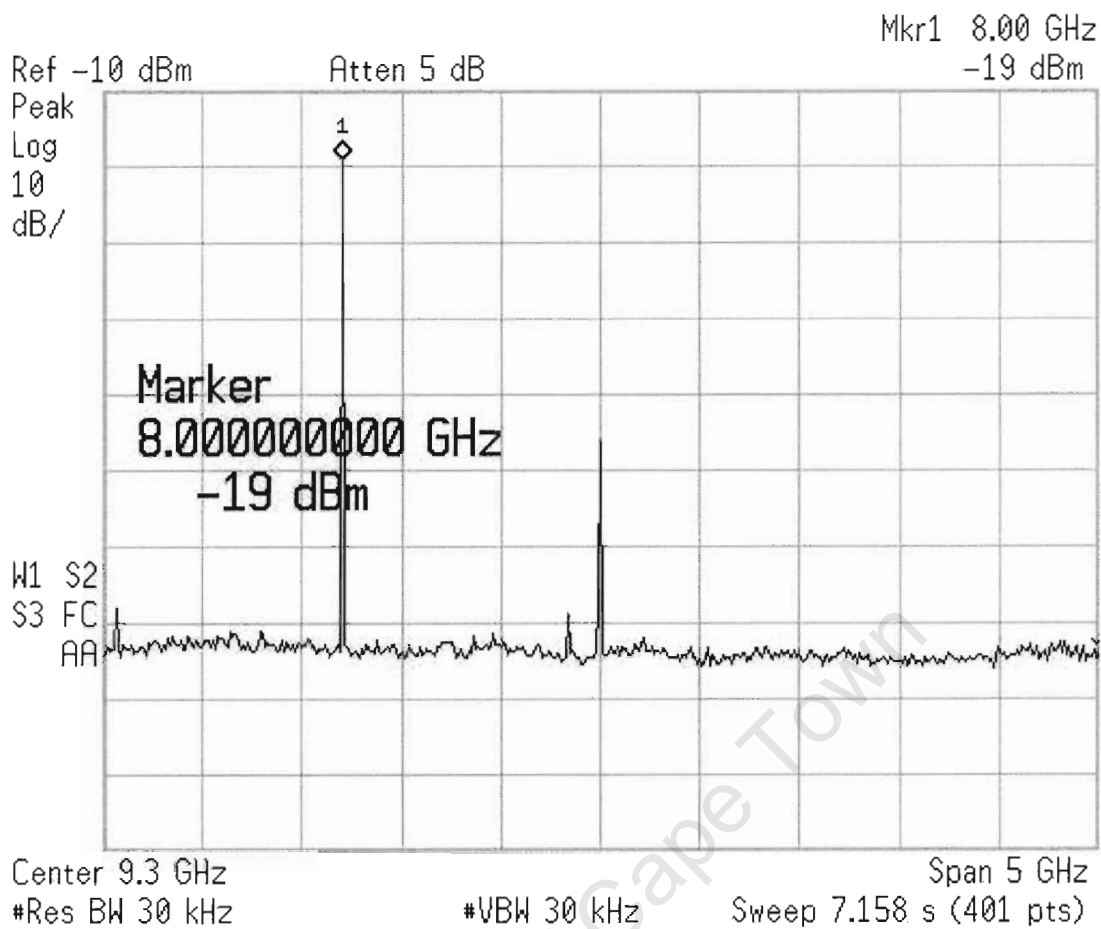


Figure 5.18: Spectrum without the presence of the 9300 MHz chirp.

5.8 System Transfer Function

5.8.1 Equipment Used

DC power supply, spectrum analyzer, AWG.

5.8.2 Test Procedure

The first IF section is connected (with appropriate FDU LO) and DC supply attached. The synthesizer is programmed and the DPG applied to the input. The resultant chirp signal is then recorded on the spectrum analyzer. The entire system is then inserted and relevant DC supply attached and synthesizers programmed. A record is made of the X-band chirp on the spectrum analyzer.

5.8.3 Test Results

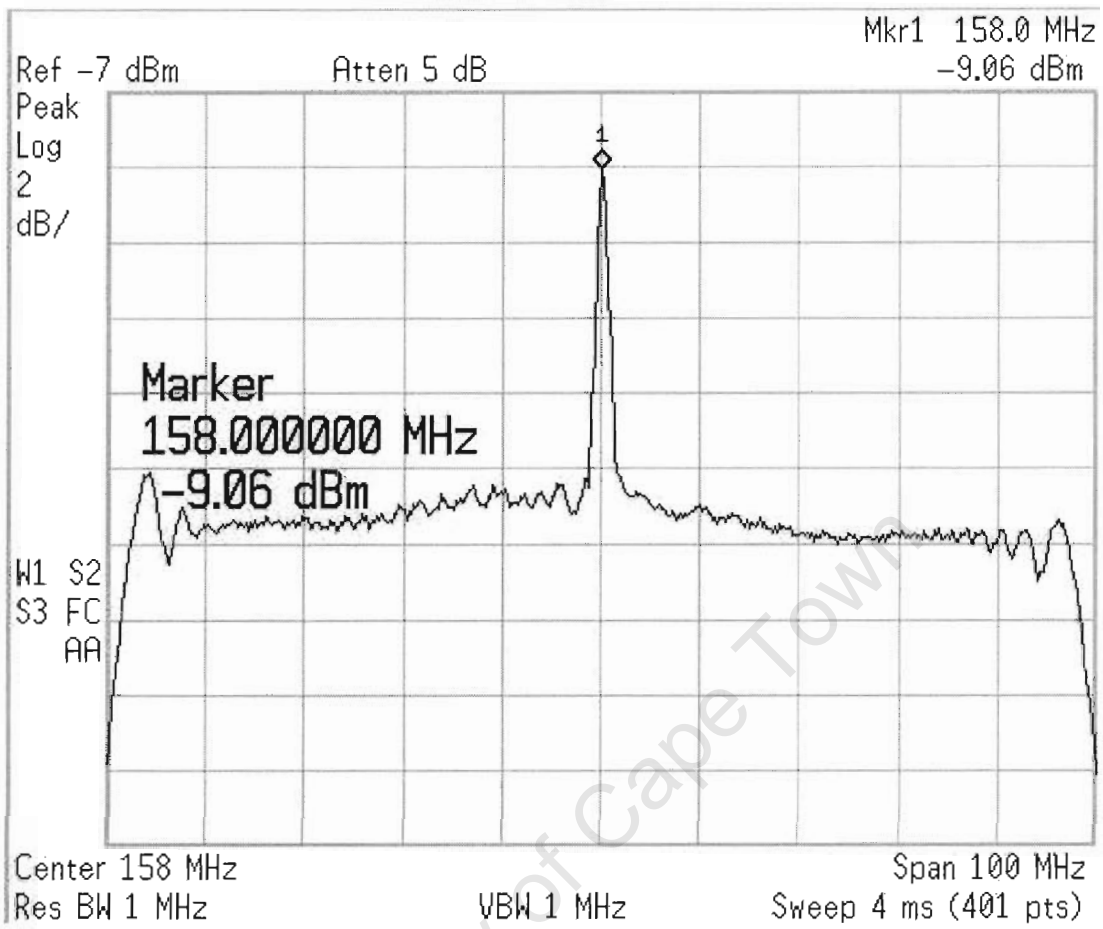


Figure 5.19: Spectrum of 158 MHz chirp.

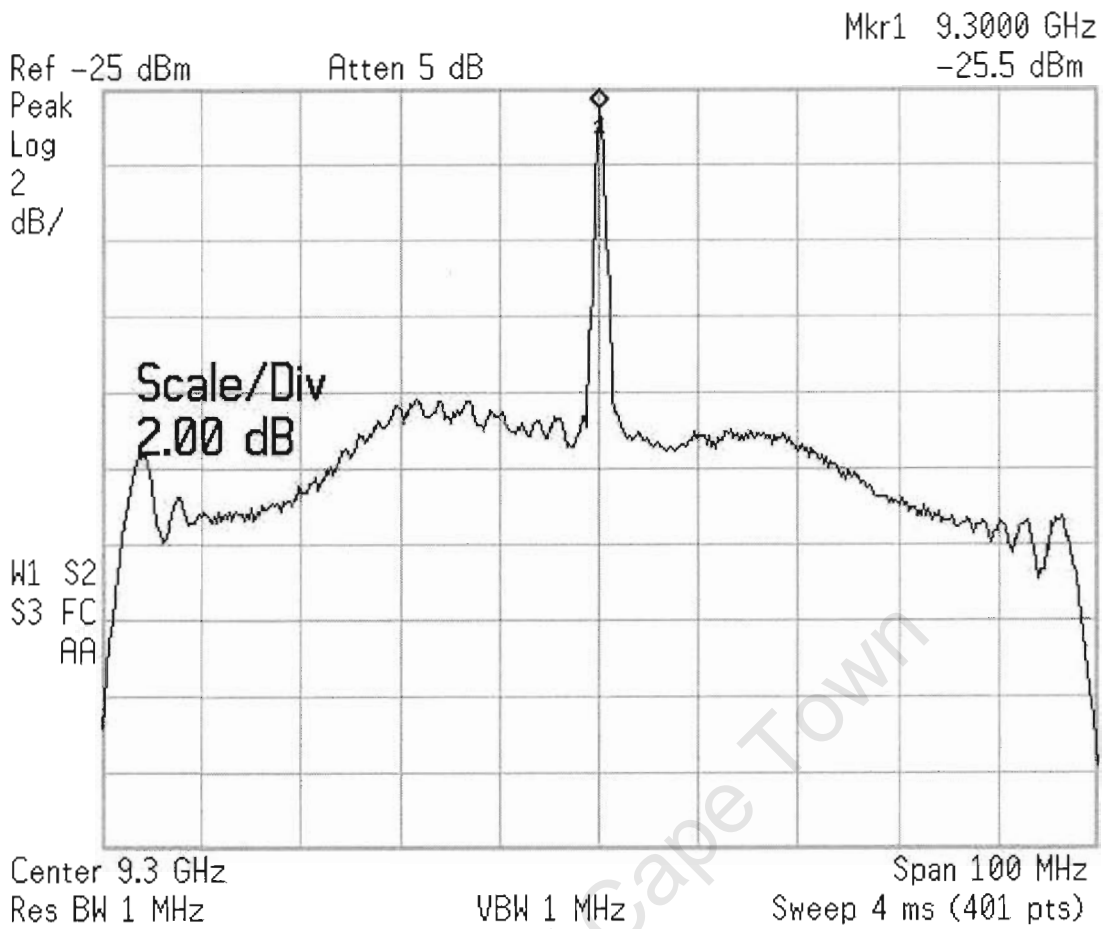


Figure 5.20: Spectrum of 9300 MHz chirp.

5.9 Discussion and Analysis of Findings

5.9.1 Amplifier Testing

The table below shows the measured and rated gain of the FDU amplifiers.

Table 5.7: Test results for the various amplifier boards.

Amplifier name	Frequency (MHz)	Measured Gain (dB)	Rated Gain (dB)
Amp11-2	158	10.5	11.0
Amp11-3	158	10.4	11.0
Amp12-2	1142	13.4	13.0
Amp12-3	1142	11.2	11.0
Amp13-2	4000	11.0	11.0
Amp13-3	8000	15.8	16.0
Amp13-4	8000	15.8	16.0

On the whole, the FDU amplifiers realise their rating without much deviation. It is also

seen that the transmitter amplifier does not deviate by more than 0.7 dB in terms of magnitude linearity over the system bandwidth.

5.9.2 Synthesizer Testing

All the five synthesizers generated their required frequency with error less than 2 kHz. Table 5.8 shows that their power specifications are also met.

Table 5.8: Measured synthesizer output power level compared with rated value.

Synthesizer name/frequency (MHz)	Measured Output (dBm)	Rated Output (dBm)
SZ1/158	+5.42	+5.0±2
SZ2/1142	+4.70	+3.0 (min)
SZ3/4000	+3.08	+3.0±2
SZ4/150	+5.27	+5.0±2
SZ5/210	+5.43	+5.0±2

The output frequency should reflect the reference frequency within the loop bandwidth. Outside the loop bandwidth, the properties of the oscillator itself dominate. The step size and noise of associated circuitry also contribute.

An applet by Raltron Electronics Corporation [20] is used to convert the phase noise plots to jitter values. These are stated in the Table 5.9 below.

Table 5.9: Calculated jitter specifications for the synthesizers.

Synthesizer name/frequency (MHz)	Jitter (ps)
SZ1/158	11.1
SZ2/1142	2.86
SZ3/4000	2.15
SZ4/150	12.7
SZ5/210	10.4

What must be realised here is that the intended reference oscillator was not available at the time of testing. The phase noise plots were produced using the AWG as the reference. Therefore, these results are not a true reflection of the final system. They are, however, included to give an indication of the jitter. The 150 and 210 MHz synthesizer jitter values are crucial as stated in the Terms of Reference (Section 1.1). Since the GPS crystal oscillator is known to be superior in terms of stability, these values show promise, as they are not too far off the required 6 ps.

5.9.3 Power Levels of the FDU

The power levels at the FDU outputs are compared to those needed at clock and mixer LO inputs in Table 5.10. All levels exceed those required. The 158 MHz arm may be padded down with a 6 dB attenuator, while a 3 dB attenuator will suffice for the receive side of the 1142 MHz arm.

Table 5.10: Measured power compared to the required minimum for the FDU.

Arm frequency (MHz)		Final Output (dBm)	Minimum Required Power (dBm)
158	I	+13.7	+7.0
	Q	+13.6	+7.0
1142	T	+12.2	+10.0
	R	+12.6	+7.0
4000	T	+12.1	+8.0
	R	+12.1	+8.0
150	-	+5.27	+4.0
210	-	+5.43	+4.0

5.9.4 Power Levels of the Transmitter

The transmit power budget, Table 5.11, confirms the calculated power levels of the Fig 3.16. The test cables and connectors will account for the additional losses. With the addition of the transmitter amplification, the power will rise to 16.5 dBm. A 6 dB attenuator can then bring this down to the required +10 dBm. Alternatively, the amplification on the DPG outputs can be adjusted accordingly.

Table 5.11: Measured versus calculated transmitter power levels.

	Input Power (dBm)	IF1 Power (dBm)	IF2 Power (dBm)	RF Power (dBm)
Measured	0	-6.6	-16.8	-23.5
Calculated	0	-6.1	-15.2	-21.9

5.9.5 Chirp Integrity

The chirp signal delivered by the DPG is of satisfactory quality. However, the power level is approximately 25 dB below that stated in the Terms of Reference and 15 dB below what the transmitter input can tolerate. It will therefore need amplification in order for the TWT to receive its intended power requirement.

Combination of the I and Q baseband signals is achieved by the process detailed in Section 3.1. There is evidence of LO feedthrough from the two 158 MHz mixers. The chirp seen

at the second IF (Fig. 5.11) shows evidence of attenuation at the outer frequencies. This has resulted from the filter bandwidth (120 MHz) not being large enough to combat the effects of group delay. This is carried through to the RF signal, which does not suffer the same fate.

5.9.6 Spurious Signals

With the video/resolution bandwidths decreased, the noise floor drops accordingly. This results in the unmasking of previously unseen frequency components, much like sprouting grey hairs. Fig. 5.13 shows the LO feedthrough and 316 MHz LO harmonic clearly. Without the chirp present in Fig. 5.14 things are less clear. The only possible source of these signals would be the DPG, since the LO has been characterised.

The LO harmonic of the previous stage disappears in the 1300 MHz spectrum. The 1142 MHz LO feedthrough still exists though, despite the efforts of the image rejection mixer. Its effort, together with the filter concerned, do provide in excess of 80 dB isolation. The image, however, is attenuated below the noise floor. The presence of the two spikes offset by 16 MHz from the 1300 MHz signal is a result of the previous stage. Once again, this trend continues through to the X-band spectrum. As would be expected, due to the lack of a filter, harmonics at 8000 MHz spacings exist. It must be noted that the video and resolution bandwidths were increased in Fig. 5.18 to cover the the 5000 MHz span.

5.9.7 System Transfer Function

Fig. 5.21 indicates how the chirp envelope varies from IF1 output to RF output.

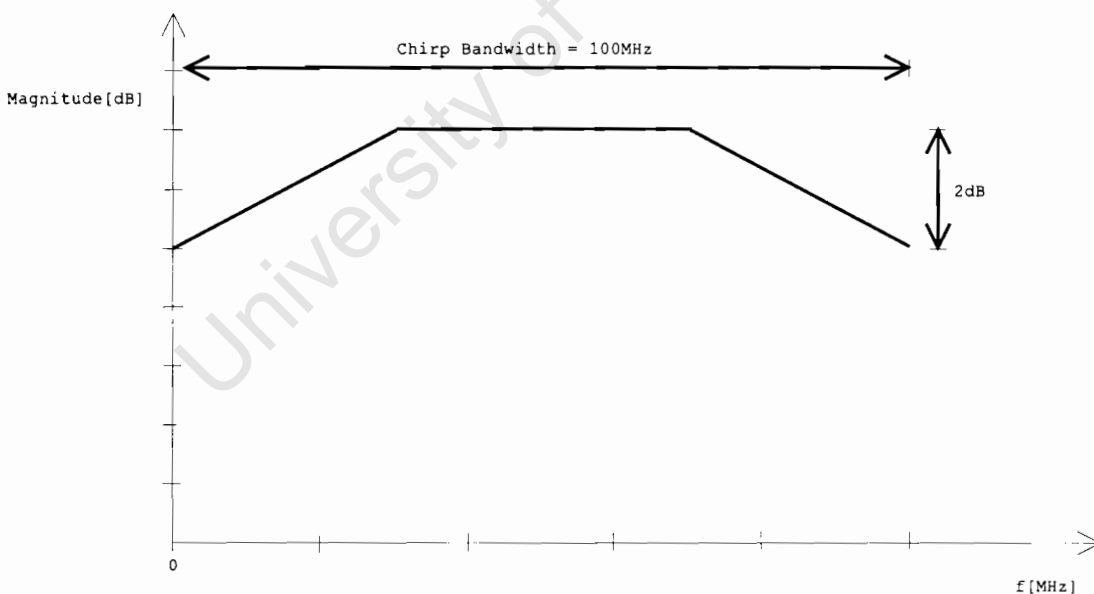


Figure 5.21: System transfer function.

Chapter 6

Conclusions and Recommendations

6.1 Conclusions

In view of the findings, the following conclusions may be drawn:

- All the FDU amplifiers provide sufficient gain for their application.
- The synthesizers output their desired frequency at sufficient power level.
- It is not clear if the jitter specification of 6 ps has been met, although it does seem promising.
- The FDU supplies sufficient power at all of its outputs.
- The DPG was found to be lacking of 24 dB. Transmitter power levels will supply the desired +10 dBm to the TWT if this is corrected for.
- Combination of the in-phase and quadrature chirp signals is done satisfactorily. LO feedthrough, however, is superimposed at the chirps' centre frequency.
- As a result of previous stages spurious signals exist at 16 MHz offset from the chirps' centre frequency and at 9142 MHz. Harmonic intermodulation products of the final mixer are also present.
- The system transfer function reveals that 2 dB attenuation is present at the outer frequencies of the chirp signal. Group delay in the filters and transmitter amplifiers is held responsible for this.

6.2 Recommendations for Future Work

As a result of the findings and conclusions, the following recommendations are made:

- The frequency synthesizers' jitter specification should be calculated when the GPS reference is acquired.
- Amplification of at least 15 dB must be performed to the DPGs' outputs.
- The IF1 mixers could be replaced with higher demands on LO-RF isolation.
- Analysis may be done to determine the origin of the 16 MHz offset spurious responses.
- The TWT must be inserted to test the need for X-band filtering.

University of Cape Town

Appendix A

Matlab Chirp Simulation Code

```
Fs = 630e6;           % Sample Rate
Tp = 5e-6;           % Chirp pulse length in seconds
ChirpBW = 100e6;     % Chirp Bandwidth

NumSamples = round(Fs * Tp);
I = zeros(1,NumSamples);
Q = zeros(1,NumSamples);
Tplot = zeros(1,NumSamples);
TimePerSample = 1/Fs;
Gamma = ChirpBW/Tp;

%%%% Create Chirp %%%
for count = 0:(NumSamples-1)
    Time = (TimePerSample*count-(Tp/2));
    Phase = pi * Gamma * Time^2;
    I(1,count+1) = cos(Phase);
    Q(1,count+1) = sin(Phase);
    Tplot(1,count+1) = (Time+(Tp/2));
end

%%%% Take FFT of chirp %%%
Chirp = complex(I,Q);
ChirpFFT = fft(Chirp);

%%%% modulation section %%%
Fcarrier = 157.5e6;   % Modulation Frequency
N = 10;              % Number of address bits
dF = Fs * 2^(-N);    % Frequency resolution
FCN = round(Fcarrier/dF); % Frequency Control Number
NCOSin = zeros(1,NumSamples);
NCOCos = zeros(1,NumSamples);
for count = 1:NumSamples
    NCOCos(count) = cos(2*pi*(mod(count*FCN,2^N))/2^N);
    NCOSin(count) = sin(2*pi*(mod(count*FCN,2^N))/2^N);
end
IMod = NCOCos.*I;
QMod = NCOSin.*Q;
FIMod = fft(IMod);
FQMod = fft(QMod);
ChirpMod = IMod + QMod;
ChirpModFFT = fft(ChirpMod);
NCOSinFFT = fft(NCOSin);

%%%% Shift FFT %%%
ChirpModFFT = fftshift(ChirpModFFT);
NCOSinFFT = fftshift(NCOSinFFT);
ChirpFFT = fftshift(ChirpFFT);
```

```
%%%%%%%%%%%% plot graphs %%%%%%%%%%%%%%%
```

```
step = Fs/(NumSamples*1e6);  
x = -Fs/(2*1e6):step:FsWithstep;  
figure,plot(x,abs(FIMod))  
xlabel('Frequency [MHz]')  
ylabel('Magnitude')  
title('I Channel at 158MHz IF');  
figure,plot(x,abs(FQMod))  
xlabel('Frequency [MHz]')  
ylabel('Magnitude')  
title('Q Channel at 158MHz IF');  
figure,plot(x,abs(ChirpModFFT))  
xlabel('Frequency [MHz]')  
ylabel('Magnitude')  
title('Chirp at 158MHz IF');  
figure,plot(x,abs(NCOSinFFT))  
xlabel('Frequency [MHz]')  
ylabel('Magnitude')  
title('158MHz sinusoid')  
figure,plot((Tplot/1e-6),I);  
xlabel('Time [us]')  
ylabel('Magnitude')  
title('I Channel')  
figure,plot((Tplot/1e-6),Q);  
xlabel('Time [us]')  
ylabel('Magnitude')  
title('Q Channel')  
end  
  
end
```

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Appendix B

Synthesizer Plots

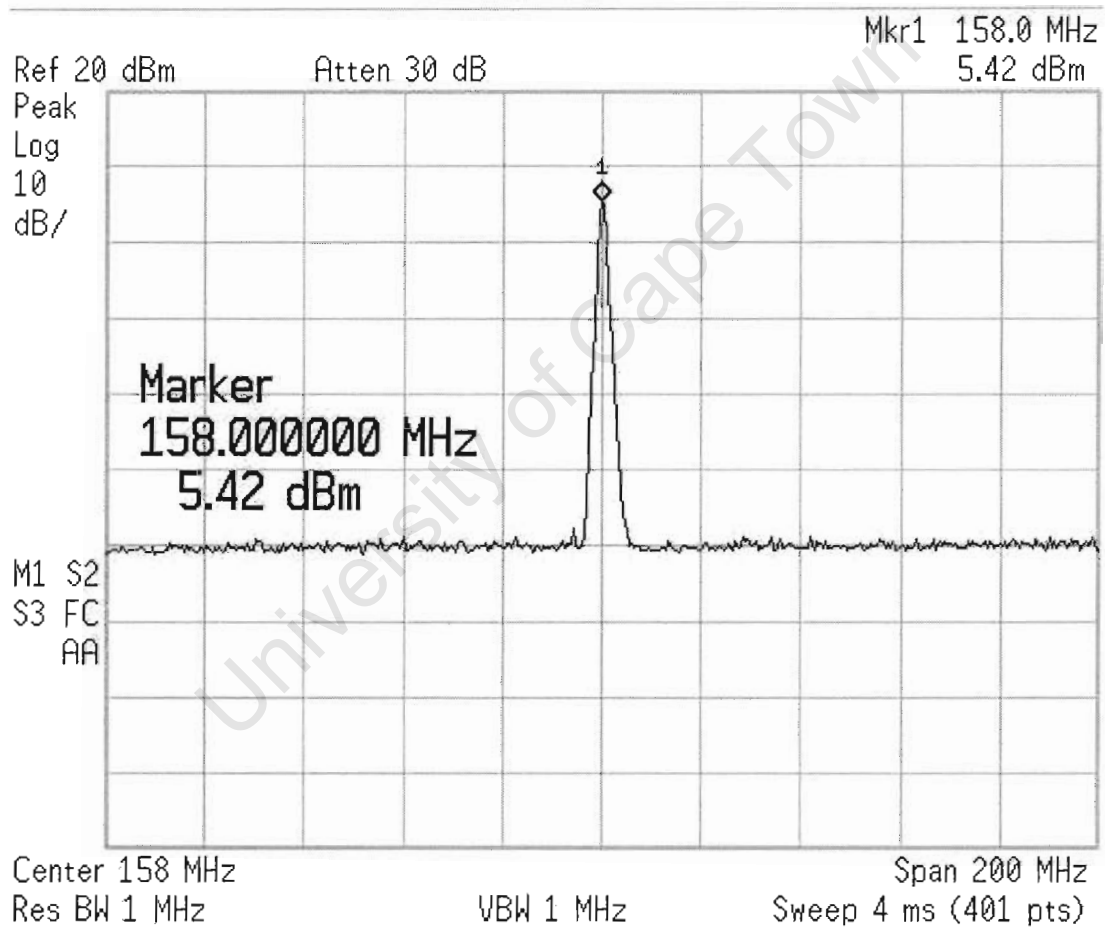


Figure B.1: Plot showing frequency and power for the 158 MHz synthesizer.



Figure B.2: Plot showing harmonics of the 158 MHz synthesizer.

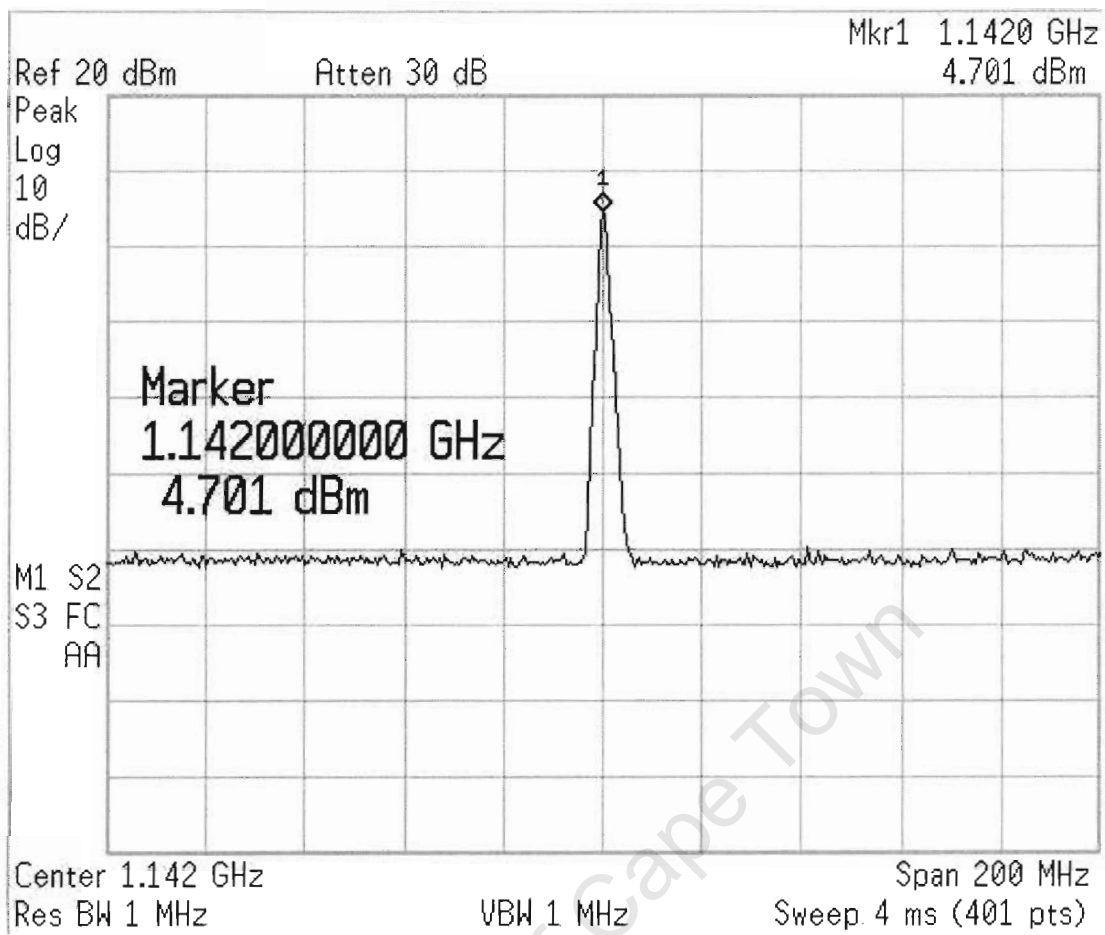


Figure B.3: Plot showing frequency and power for the 1142 MHz synthesizer.



Figure B.4: Plot showing harmonics of the 1142 MHz synthesizer.

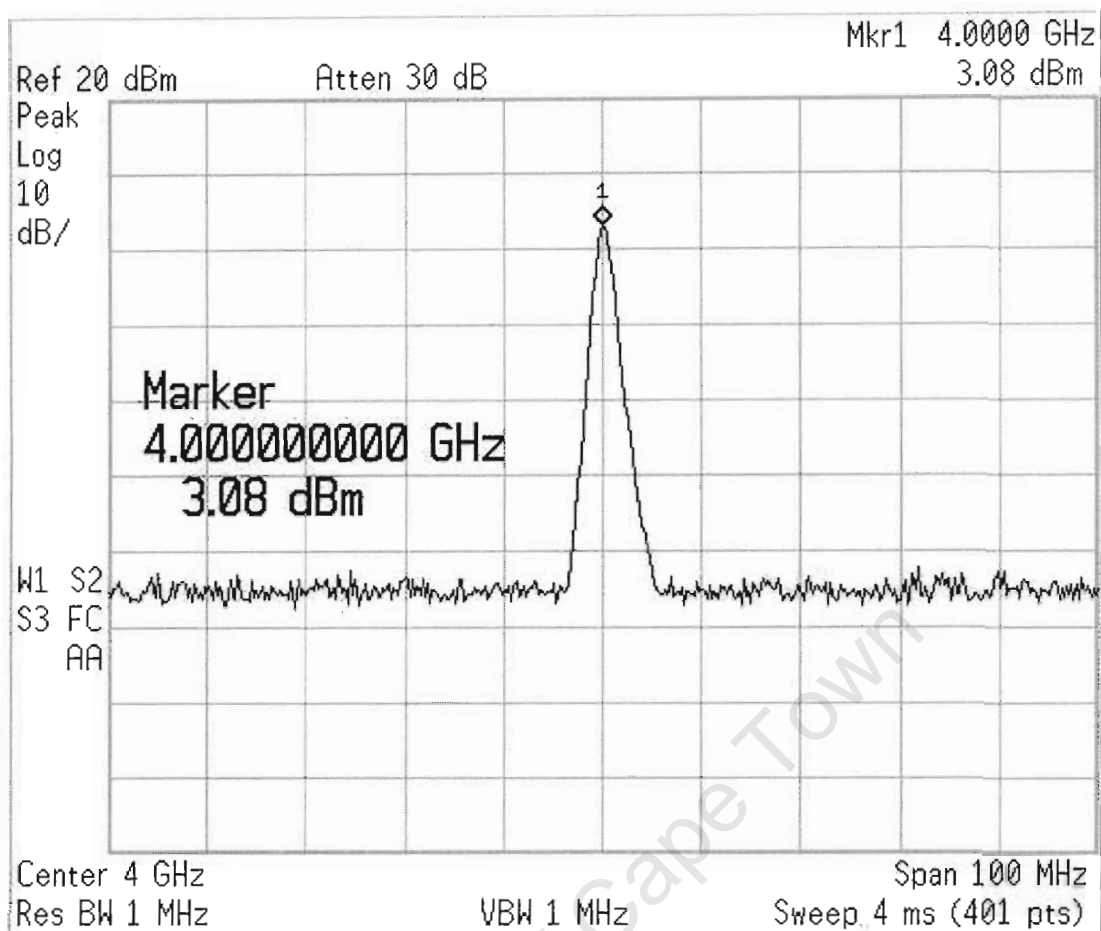


Figure B.5: Plot showing frequency and power for the 4000 MHz synthesizer.

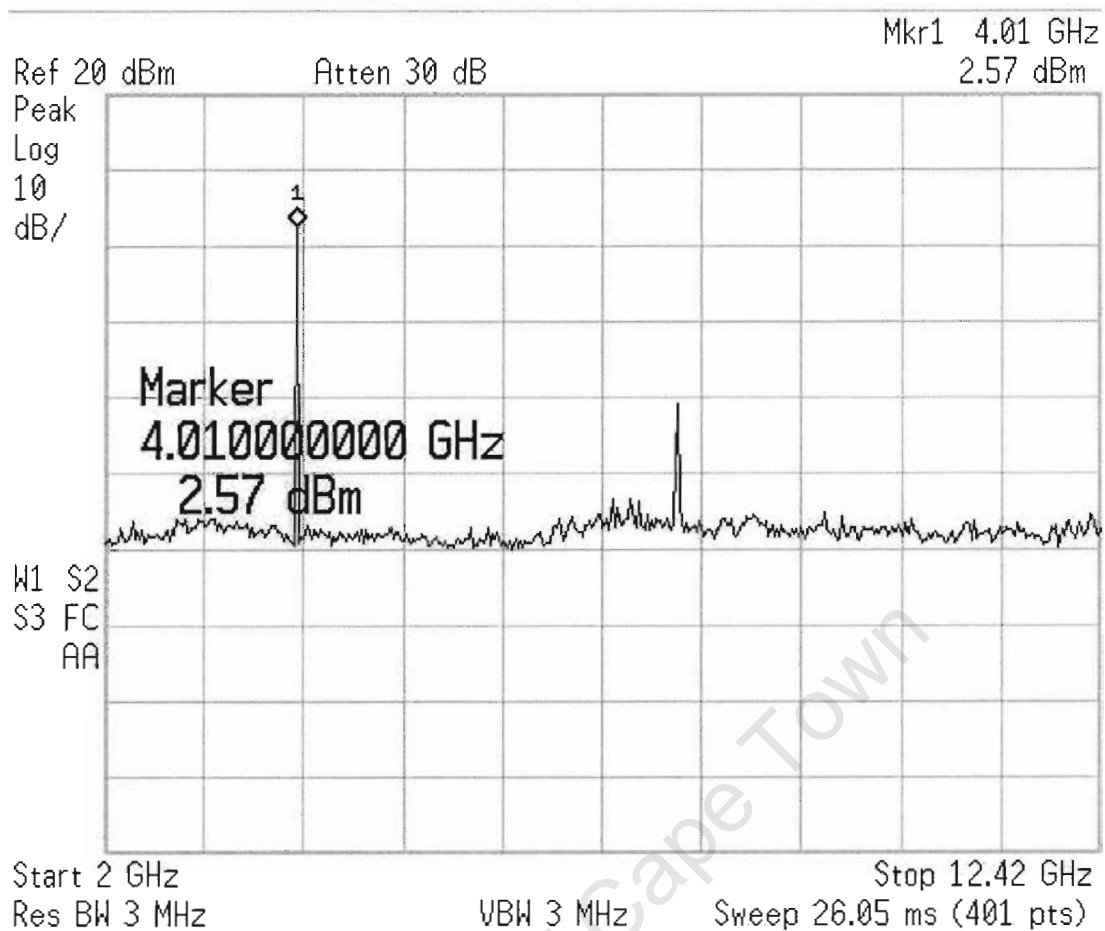


Figure B.6: Plot showing harmonics of the 4000 MHz synthesizer.

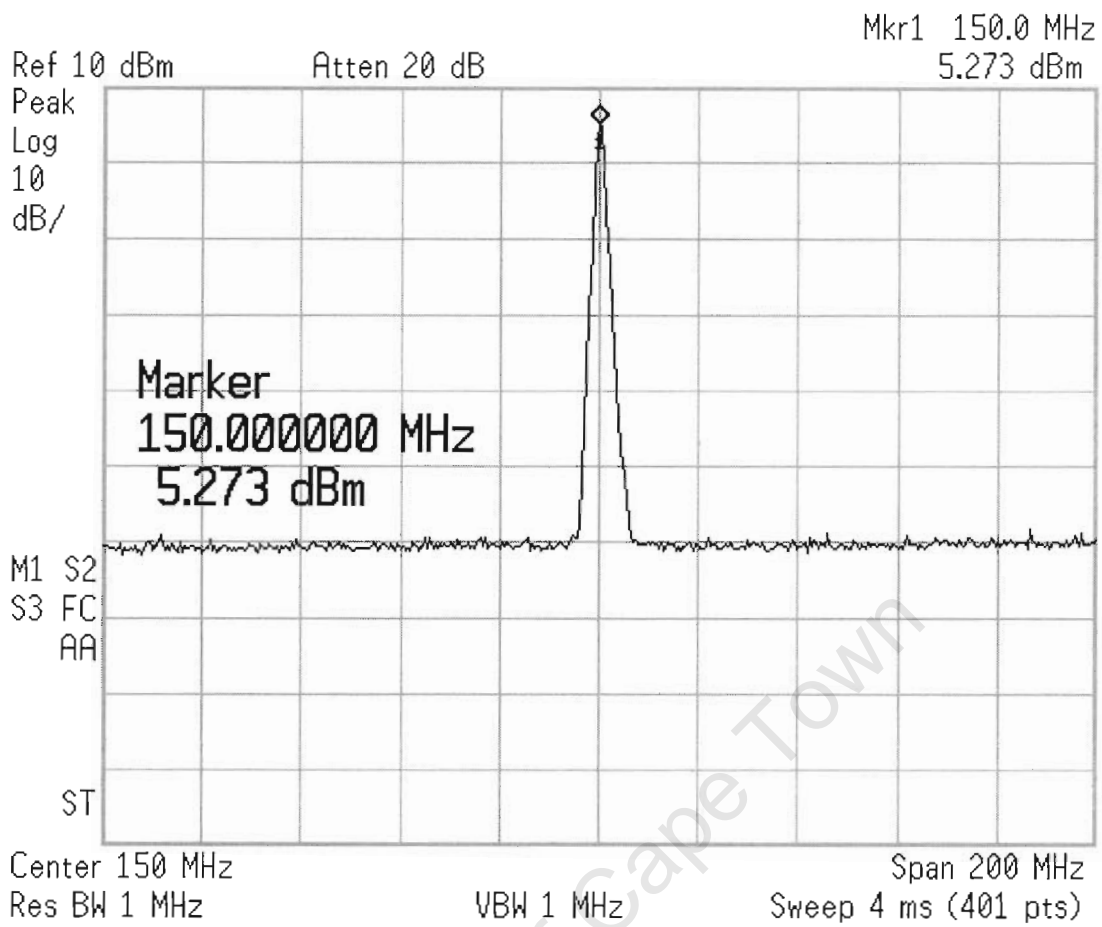


Figure B.7: Plot showing frequency and power for the 150 MHz synthesizer.

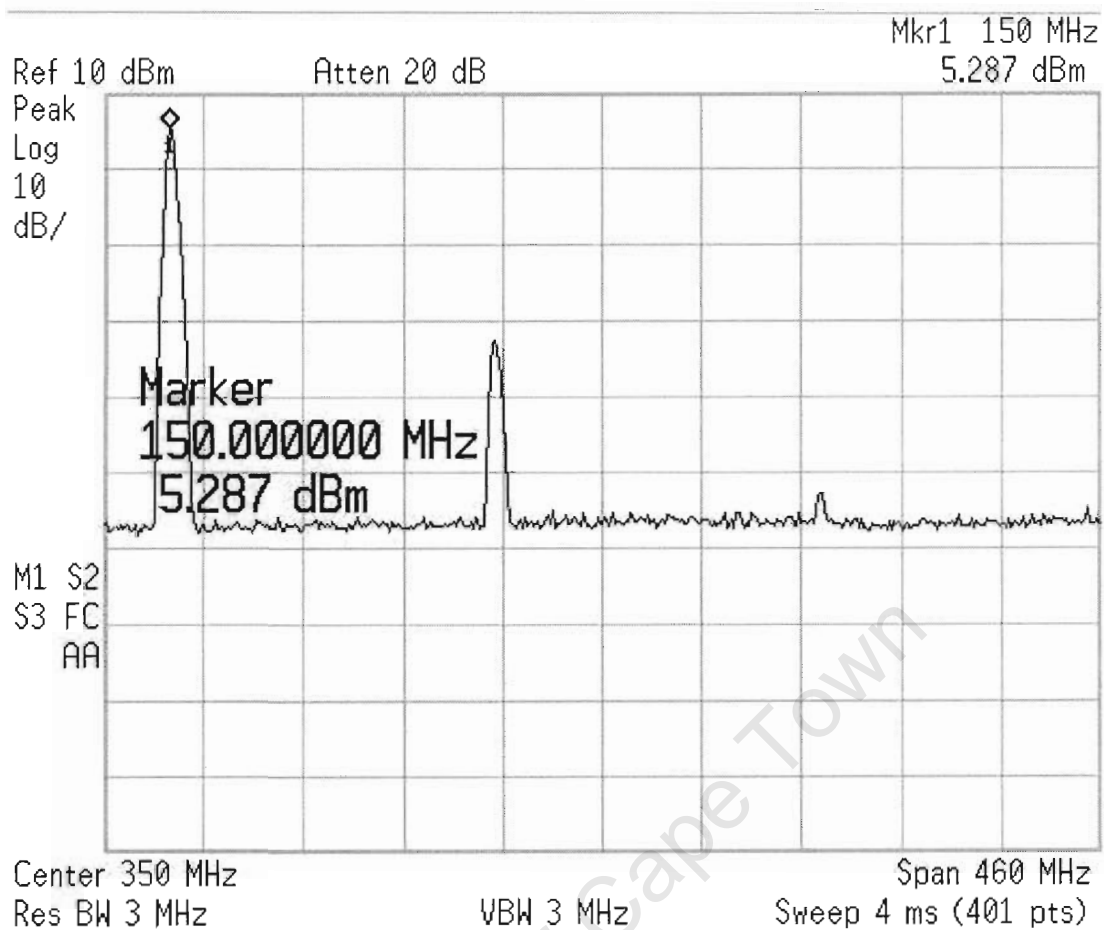


Figure B.8: Plot showing harmonics of the 150 MHz synthesizer.

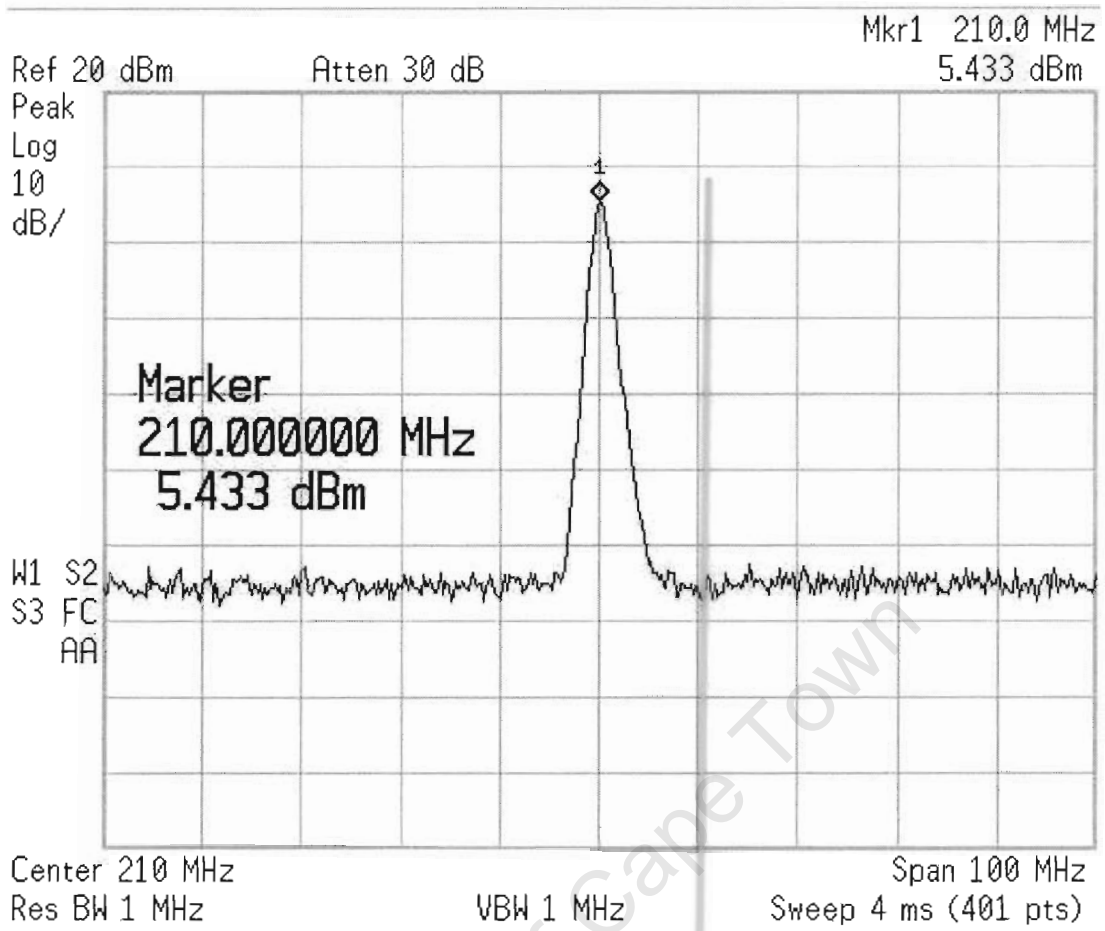


Figure B.9: Plot showing frequency and power for the 210 MHz synthesizer.

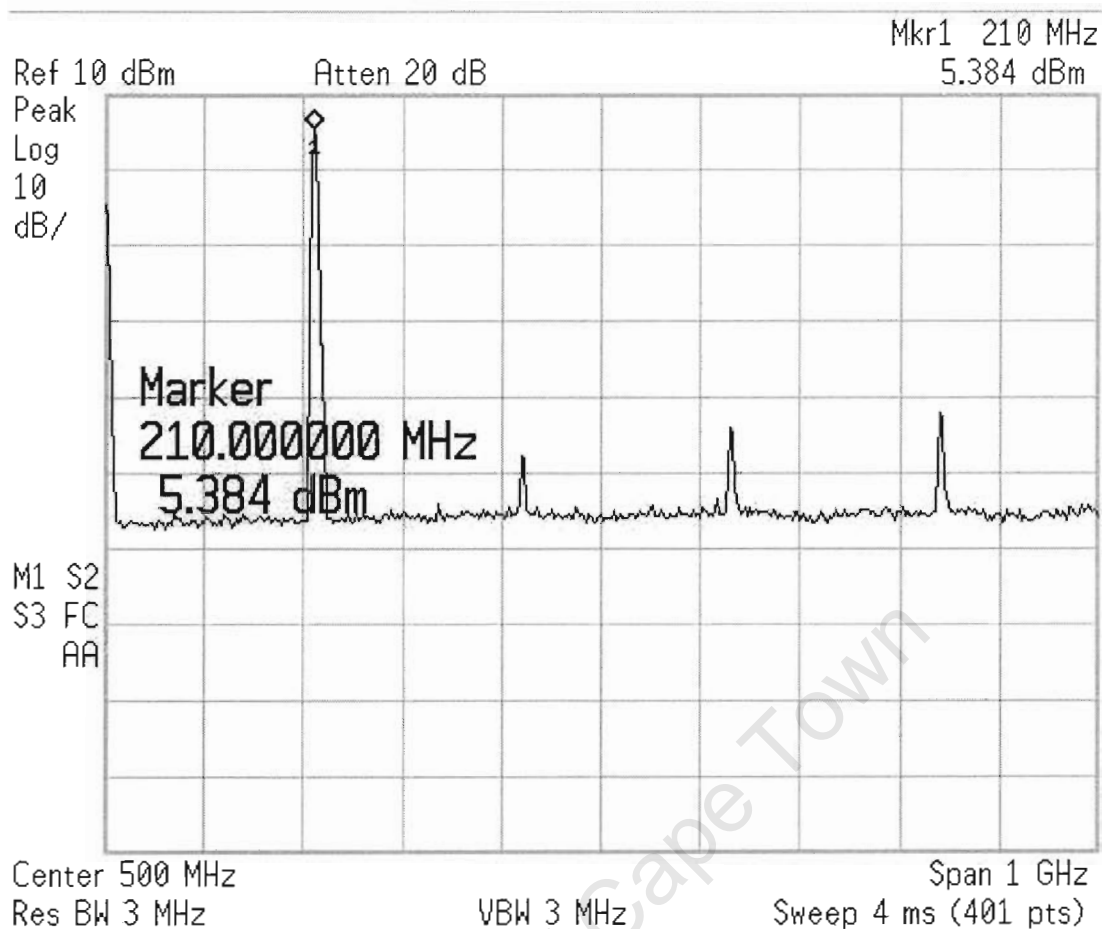


Figure B.10: Plot showing harmonics of the 210 MHz synthesizer.

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